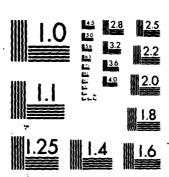
NAVAL POSTGRADUATE SCHOOL MONTEREY CA F/6 18/11 AN INTERACTIVE CODE FOR A PRESSURIZED WATER REACTOR INCORPORATI==ETC(U) JUN 80 6 6 MEATH AD-A092 304 NL UNCLASSIFIED Albert a



MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

AD A 0 9 2 3 0 4

BUC FILE COPY

LEVEL



NAVAL POSTGRADUATE SCHOOL

Monterey, California





E

THESIS

AN INTERACTIVE CODE FOR A PRESSURIZED WATER REACTOR INCORPORATING TEMPERATURE AND XENON FEEDBACK

bу

Gregory Garver Heath

June 1980

Thesis Advisor:

P. J. Marto

Approved for public release; distribution unlimited

Unclassified

REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
1,	ON MO. 3. RECIPIENT'S CATALOG NUMBER
AD-A093 3	S. TYPE OF REPORT A PERIOD COVERED
An Interactive Code for a Pressurized Water (Master's Thesis June 1980
Reactor Incorporating Temperature and Xenon Feedback.	6. PERFORMING ORG, REPORT NUMBER
7. AUTHORY)	S. CONTRACT OR GRANT NUMBER(s)
Gregory Garver Heath	
. PERFORMING ORGANIZATION NAME AND ADDRESS	IG. PROGRAM ELEMENT, PROJECT, TASK
Naval Postgraduate School Monterey, California 93940	(13) 11/
11 CONTROLLING OFFICE NAME AND ADDRESS	12. REPORT DATE
Naval Postgraduate School	June 1380
Monterey, California 93940	13. NUMBER OF PAGES
14. MONITORING AGENCY NAME & ACCRESS/II different from Controlling O	
Naval Postgraduate School	Unclassified
Monterey, California 93940	The DECLASSIFICATION/DOWNGRADING
Approved for public release; distribution unli	
Approved for public release; distribution unli	
Approved for public release; distribution unli	
Approved for public release; distribution unli 17. DISTRIBUTION STATEMENT (of the shortest entered in Block 20, if diffe 18. SUPPLEMENTARY NOTES	rest from Report)
Approved for public release; distribution unli	reat from Report)
Approved for public release; distribution unli 7. DISTRIBUTION STATEMENT (of the energet entered in Block 20, if difference to the supplementary notes 8. KEY WORDS (Continue on reverse side if necessary and identify by block to the supplementary notes.)	lerator Feedback lerator Feedback lerator from Shippingport Atomic tions for one delayed neutron turbation theory. The model of feedback effects, as well as tor coolant temperature control to was simulated for normal

DD 1 JAN 73 1473 (Page 1)

A CONTRACTOR OF THE STATE OF TH

201710N 07 1 NOV 68 13 0850L 8/N 0102-014-8401 : 2015/105

Unclassified

1 SECURITY CLASSIFICATION OF THIS PAGE (When Boto Smicrod)

O

BOSUMTY EL ADDIFICATION OF THIS PAGGIFFOR Rote BE

post-shutdown Xenon transient response was also modeled. The interactive program was coded in FORTRAN-IV language, and the simulation program was coded in IBM CSMP-III language.

Acces	sion For	
NTIS DDC T	GRALI	X
Unamn	ounced	H
Justi:	fication_	
Ву	·	, , , , , , , , , , , , , , , , , , , ,
Distr	lbution/	
_Avai	edifity (odes
	Avail and	•
Dist.	special	
17		

Approved for public release; distribution unlimited

An Interactive Code for a Pressurized Water Reactor Incorporating Temperature and Xenon Feedback

bу

Gregory Garver Heath
Lieutenant Commander, United States Navy
B.S.M.E., United States Naval Academy, 1971

Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN ENGINEERING SCIENCE

from the

NAVAL POSTGRADUATE SCHOOL June, 1980

Author	heaver & Newson
Approved by:	D. Marto
	Chairman, Department of Mechanical Engineering
	Chairman, Department of Mechanical Engineering
	Dean of Science and Engineering
	Dean of Science and Engineering

ABSTRACT

An interactive computer model of a highly enriched pressurized water reactor was developed, using the applicable plant parameters from the Shippingport Atomic Power Station. The point reactor kinetics equations for one delayed neutron precursor group were linearized using small perturbation theory. The model included both moderator and Xenon-135 reactivity feedback effects, as well as an automatic reactor protection and average reactor coolant temperature control system. The thermal response of the model plant was simulated for normal operating transients induced either by control rod or turbine load changes. The post shutdown Xenon transient response was also modeled. The interactive program was coded in FORTRAN-IV language, and the simulation program was coded in IBM CSMP-III language.

TABLE OF CONTENTS

I.	INTRODUCTION	9
II.	MODEL DESIGN CONSIDERATIONS	11
	A. PRESSURIZED WATER REACTOR	11
	B. MODEL AND PROGRAM FLEXIBILITY	13
III.	MODEL DEVELOPMENT	15
	A. POINT REACTOR KINETICS EQUATIONS	15
	1. Zero Power Reactor Transfer Function	17
	2. Limitations	18
	B. REACTIVITY FEEDBACK MECHANISMS	19
	1. Xenon-135 Transfer Function	21
	2. Moderator Temperature Feedback	24
	C. THERMAL ANALYSIS	26
	1. Reactor Heat Transfer Function	26
	2. Heat Exchanger Transfer Function	29
	3. Primary Piping Transfer Functions	35
	D. OVERALL PLANT BLOCK DIAGRAM	37
	E. MODEL CONSTANTS	38
IV.	RESULTS	42
	A. INTERACTIVE PROGRAM	42
	B. SIMULATION PROGRAM	48
٧.	CONCLUSIONS AND RECOMMENDATIONS	54
APP	ENDIX A: DEVELOPMENT OF THE XENON FEEDBACK TRANSFER FUNCTION COMPONENTS	56
APP	ENDIX B: DEVELOPMENT OF THE REACTOR HEAT TRANSFER	61

APPENDIX C:	DEVELOPMENT OF THE HEAT EXCHANGER TRANSFER FUNCTION	65
APPENDIX D:	DEVELOPMENT OF THE PRIMARY PIPING FUNCTIONS	67
APPENDIX E:	INTERACTIVE PROGRAM LISTING	71
APPENDIX F:	SIMULATION PROGRAM LISTING	78
APPENDIX G:	INTERFACE PROGRAM LISTING	91
LIST OF REFE	RENCES	92
INITIAL DIST	RIBUTION LIST	93

The second secon

LIST OF FIGURES

1.	Schematic of the Model Plant	12
2.	Block Diagram of the Zero Power Reactor Transfer Function	18
3.	Block Diagram of the Reactor Transfer Function with Xenon-135 and Moderator Temperature Feedback Loops, and External Reactivity	20
4.	Xe-135 Decay Scheme	21
5.	Simplified Xenon-135 Decay Scheme	22
6.	Block Diagram of the Reactor Transfer Function with Xenon-135 Feedback and External Reactivity	24
7.	Block Diagram of the Reactor Heat Transfer Function	29
8.	Block Diagram of the Heat Exchanger Transfer Function	34
9.	Block Diagram of the Transport and Mixing Delay Functions	38
10.	Overall Plant Block Diagram	39
11.	Interactive Program, Introduction and Instructions	43
12.	Interactive Program, Initial Power Level and Simulation Choice Inputs	44
13.	Post-shutdown Xenon-135 Behavior	45
14.	Interactive Program, Control Rod Movement Inputs	46
15.	Interactive Program, Turbine Load Change Inputs	47
16.	Interactive Program, Logic Routine Example	49
17.	Power, Reactivity, and Temperature Response to a 20 to 40% Turbine Load Change	50
18.	Power, Reactivity, and Temperature Response to a 60 to 40% Turbine Load Change	51
19.	Power, Reactivity, and Temperature Response to a	52

20.	Power, Reactivity, and Temperature Response to a 10 second Outward Control Rod Movement	53
D-1.	Mixing Volume	69

I. INTRODUCTION

Computers have been used in the design and analysis of nuclear reactors since the inception of reactor technology, and many codes have been developed. More recently, digital computer programs have been used as learning devices for nuclear engineering students. The formulation of the computational problem of predicting the behavior of a nuclear reactor has been greatly facilitated with the use of high level computer languages.

The purpose of this work was to develop a computer assisted learning device to be used by students taking nuclear engineering courses at the Naval Postgraduate School. The program graphically displays the simulated kinetic and thermal transient responses of a pressurized water reactor power plant. The reactivity feedback effects of Xenon-135 poisoning and moderator temperature are incorporated into the model. Normal operating transients, starting from a steady state critical condition, can be induced by ramp changes in either control rod position or turbine load. The initial reactor power level and reactivity change mechanism are chosen by the program user. For user convenience, these inputs are prompted and entered interactively, after which the simulation is run with no further user action required.

A pressurized water reactor (PWR) was modeled as it is the most common type of reactor used for power plant application. As a realistic reference, the model design incorporated the applicable characteristics of the Shippingport Atomic Power Station, whose nuclear parameters were the most compatible, of available PWR core data, with the assumption of a highly enriched core made in the model design.

The model was developed by treating the reactor kinetics and other plant component thermodynamic relationships as transfer functions [1]. The point reactor kinetics equations were linearized using a Taylor expansion or small perturbation technique [2]. This same perturbation technique and lumped parameter analysis were applied to the plant's linear differential heat transfer and Xenon-135 equations [3]. The individual transfer functions were then integrated into an overall plant block diagram. The program also features a reactor protection and average reactor coolant temperature control scheme.

IBM CSMP-III language was chosen to formulate the model simulation as it has inherent routines to invert the transfer functions to the time domain.

II. MODEL DESIGN CONSIDERATIONS

Lumped parameter analysis and first order perturbation theory were used throughout the model development. All the nuclear variables which were used in the reactor kinetics and Xenon-135 decay equations were considered to be averaged values of the variables over the neutron energy spectrum. Similarly, the thermodynamic variables which were used in the plant's heat transfer equations were considered to be averaged values over the volume of the particular component.

Because of these approximations and other assumptions made in the model development, the simulation should not be considered a design or stability analysis. Instead, the simulation shows the model's large scale reactor kinetic and thermal trends during normal operating transients.

For similar reasons, although certain plant parameters of the Shippingport Atomic Power Station were incorporated into the model, the simulation cannot be considered to reflect the operating characteristics of this power plant. The Shippingport core is a seed and blanket type whereas the model core is a uniform mixture of highly enriched fuel and support materials.

A. PRESSURIZED WATER REACTOR

The pressurized water reactor (PWR) is the most widely used reactor type in central power plant applications and the only type currently used in naval propulsion.

A simplified schematic of the modeled PWR plant is shown in Figure

1. The modeled reactor contains a highly enriched core which is light

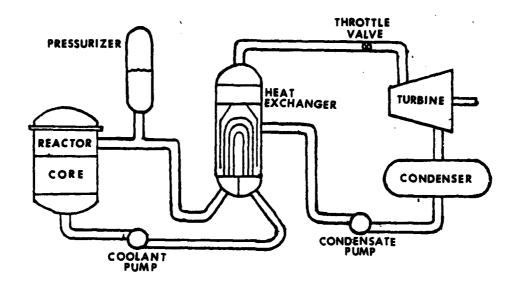


Figure 1. Schematic of the Model Plant

water moderated and cooled. The modeled plant contains two closed loop thermodynamic systems coupled by a heat exchanger.

In the primary system, heat generated from thermal fission is transferred to the coolant as it passes through the reactor, raising the coolant's temperature. The high temperature coolant leaving the reactor is circulated by a pump through tubes inside of a heat exchanger where it gives up some heat to a secondary fluid. A pressurizer maintains the primary coolant at a sufficiently high pressure so that the bulk temperature of the coolant is kept below the saturation temperature.

On the secondary side of the heat exchanger, the temperature of the entering feedwater is raised to the boiling point and saturated steam is produced. The steam is delivered to a turbine, is condensed and returned to the heat exchanger, completing the second closed loop.

B. MODEL AND PROGRAM FLEXIBILITY

The model was designed to simulate only normal operating transients from an initial steady state condition. Reactor accidents and startup were not considered in the model development. However, the model's reactor control module will simulate either a full or constant insertion of control rods if certain parameters are exceeded. This feature was not incorporated for accident analysis, but to keep parameters within the limits of the assumptions made in the model development.

The applicable Shippingport plant characteristics are fixed in the simulation program. User inputs are limited to choosing initial power level, from which other initial parameters are adjusted, and the plant perturbation mechanism, either control rod movement or turbine load change. These perturbations occur at fixed rates which limit the amount of reactivity that can be inserted during the simulation.

Originally it was envisioned that an overall program would feature not only interactive input capability but also produce real time graphical displays of the transient for a user at a time sharing computer terminal which had graphics capability. However, the computer language (IBM CSMP-III) required to solve the model's algorithms was not available on the time sharing system, and furthermore, the programs long execution time precluded any real time response.

In order to still provide as much user facility as possible with these restrictions, the interactive feature was partially retained for data input. Using a computer language available on the time sharing system (FORTRAN-IV), a program which interactively prompts the user to

enter specified controlling parameters was developed. This program incorporates logic routines which permit inputs only within requested ranges. This feature allows for data reentry if a user error is made, and ensures that only inputs compatible with the model development's assumptions are used in the simulation.

After the input is completed, the initial transient conditions are displayed at the terminal. A separate internal control program then transfers the user supplied inputs into the simulation program. The control program then transfers the simulation program to the batch processing system for execution. The interactive program notifies the user that this has been done. Hard copy plots of the transient are subsequently produced. The post-shutdown Xenon behavior module is located in the interactive program and produces real time graphical displays at the terminal.

Thus there are three distinct programs:

401 A . M. MONEY 25

- 1. An interactive program to prompt and receive user input, and also simulate post-shutdown Xenon behavior.
- 2. A batch processed program which simulates the transient response to the user inputs.
- 3. An internal control program which interfaces these two programs.

III. MODEL DEVELOPMENT

A. POINT REACTOR KINETICS EQUATIONS

A lengthly and formal derivation of the point kinetics equations is found in Reference 4 and will not be repeated here. More generally these equations are obtained from one group neutron diffusion theory with the assumption that the neutron flux is deparable in time and space, and with the inclusion of delayed neutrons [5]. They are listed below.

$$\frac{d\vec{p}(t)}{dt} = \frac{\cancel{p}(t) - \cancel{B}}{\cancel{A}} \vec{p}(t) + \sum_{i} \lambda_{i} c_{i}(t) + Q(t)$$
 (1)

$$\frac{dC_{i}(t)}{dt} = \frac{\overline{\beta}_{i}}{\Lambda} \beta(t) - \lambda_{i}C_{i}(t) \quad i = 1, 2, \dots$$
 (2)

where $\emptyset(t)$ = Flux amplitude function

 $\rho(t)$ = Reactivity

Q(t) = Extraneous delayed neutron source strength

= Effective delayed neutron fraction from the ith group

λ_i = Decay constant of the ith group

and $\vec{\beta} = \sum_{i} \vec{\beta}_{i}$ = Total effective delayed neutron fraction

These equations are referred to as the point reactor kinetics equations not because the reactor is considered as a single point in their application, but since spatial variations are neglected. $\bar{\beta}_i, \Lambda$, and λ_i are assumed to be constant.

If the neutron flux spatial variation is assumed to be time invariant, $\emptyset(t)$ can be considered to represent the number of neutrons in the

core. Equation (1) may then be seen as a neutron rate equation where the three terms on the right hand side represent the rate of production of prompt, delayed, and source neutrons respectively.

Equation (2) is a rate equation for the ith delayed neutron precursor group. The two terms on the right hand side represent production and decay rates respectively. The precursors are comprised of approximately thirty isotopes which historically have been divided into six groups with decay constants ranging from 0.0124 to 3.0 seconds ⁻¹ for ²³⁵U. In the model, the precursors were considered to be represented by one effective group characterized by an averaged decay constant [2].

$$\lambda = \left[\frac{1}{B} \sum_{i=1}^{6} \frac{\vec{\beta}_{i}}{\lambda_{i}} \right]^{-1}$$

and an effective delayed neutron fraction

$$\overline{\beta} = \sum_{i=1}^{6} \overline{\beta}_{i}$$

In equations (1) and (2), reactor power may be substituted for neutron flux if the precursor concentration is modified by $C = E_f \sum_{f} C_{old}$ since

$$P = E_f \sum_f \emptyset$$

where P = Reactor power

 E_{f} = Energy released per fission

 $\Sigma_{\rm f}$ = Macroscopic fission cross section

9 = Integrated one group neutron flux

For a reactor operating at power, the source term contribution to the overall neutron population is negligible, i.e., $Q \approx 0$.

Incorporating these assumptions, equations (1) and (2) reduce to:

$$\frac{dP(t)}{dt} = \frac{\Delta(t) - \overline{\Delta}}{\Lambda} P(t) + \lambda C(t)$$
 (3)

$$\frac{dC(t)}{dt} = \frac{3}{4}P(t) - \lambda C(t)$$
 (4)

1. Zero Power Reactor Transfer Function

Consider an initially critical steady state reactor at some power level P_0 at t ≤ 0 . Equations (3) and (4) each yield:

$$C_{o} = \frac{3}{\lambda \Lambda} P_{o}$$
 (5)

Now let $P(t) = P_0 + \sqrt{P(t)}$

$$C(t) = C_0 + \delta C(t)$$

$$A(t) = A + \Delta D(t)$$
(6)

where the zero subscript denotes the initial steady state value and the delta prefix a $\underline{\text{small}}$ perturbation imposed at t=0 about this value.

Noting that ρ_0 = 0 for a critical reactor and substituting equations (5) and (6) into equations (3) and (4) yields for t \geq 0

$$\frac{d}{dt} \mathcal{S}P(t) = \frac{Po}{\Lambda} \mathcal{S}\rho(t) - \frac{\overline{\mathcal{S}}}{\Lambda} P(t) + \lambda \mathcal{S}C(t)$$
 (7)

$$\frac{d}{dt} \mathcal{F}C(t) = \frac{3}{\Lambda} \mathcal{F}P(t) - \lambda \mathcal{F}(t)$$
 (8)

where the term in equation (7) has been neglected.

Taking the Laplace transforms of equations (7) and (8):

$$s \, dP(s) = \frac{P_0}{\Lambda} d\rho(s) - \frac{\overline{\mathcal{S}}}{\Lambda} dP(s) + \lambda d^2C(s) \tag{9}$$

$$SC(s) = \frac{3}{\Lambda}SP(s) - \lambda SC(s) \tag{10}$$

Solving equation (10) for C(s), substituting this into equation (9), and rearranging, yields:

$$\frac{\sigma'P(s)}{\sigma'\rho(s)} = \frac{P_0}{s(\Lambda + \frac{3}{s+\Lambda})} \tag{11}$$

This is the zero power reactor transfer function, so called because the reactor is assumed to be operating at a sufficiently low enough power that no feedback effects are realized. These effects are examined in the following sections.

Equation (11) may be represented in block diagram form as shown in Figure 2.

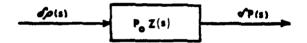


Figure 2. Block Diagram of the Zero Power Reactor Transfer Function

where
$$Z(s) = \frac{1}{S(\Lambda + \frac{3}{S+\lambda})}$$

2. <u>Limitations of the Point Reactor Kinetics Equations</u>

The point reactor kinetics equations derivation is based on the assumption that the spatial dependence of the neutron flux is negligible. This assumption limits the validity of these equations to transients where this remains a reasonable approximation such as those which result in only small changes in reactivity. The small perturbation technique used in the development of the zero power reactor

transfer function also requires only small changes in reactivity if the first order approximation is to hold. Specifically, ρ must be less than $0.5\,\vec{\beta}$. Physically, when ρ is greater than $\vec{\beta}$ the reactor is critical on prompt neutons alone. The simulated transients imposed on the model were limited to ensure that excessive amounts of reactivity were not introduced.

B. REACTIVITY FEEDBACK MECHANISMS

In the zero power reactor point reactor, the power level is assumed to be so low that it does not affect the reactivity, thus there are no feedback effects. However, for a reactor operating at a useful power level, feedback effects do exist and the reactivity becomes an implicit function of the reactor power level (or neutron flux). This dependence arises since reactivity depends on macroscopic cross sections which involve the atomic number densities of material in the core:

where Σ = Macroscopic cross section (cm⁻¹)

N = Atomic number density (atoms/cm³)

♂ = Microscopic cross section (cm [€])

The atomic number density can depend upon the reactor power level since the concentrations of certain nuclei are constantly changing due to neutron interactions. Material densities also depend upon temperature which is a function of reactor power level and hence the flux.

¹Duderstadt, J. J. and Hamilton, L. J., <u>Nuclear Reactor Analysis</u>, 1st ed., Wiley, 1976.

The two feedback mechanisms considered in the model were Xenon-let and moderator temperature.

Reactivity can also be changed directly by an external source such as control rods containing a neutron absorbing material. Thus the overall reactivity can be written as:

$$\mathcal{S}_{\rho} = \mathcal{S}_{\rho} + \mathcal{S}_{\rho} + \mathcal{S}_{\rho} + \mathcal{S}_{\rho}$$
 (12)

where

SOX = FX P

60 = Overall core reactivity

FE = Externally added reactivity

₽ = Xenon-135 feedback reactivity

Fx = Xeonon transfer function

 F_{τ} = Moderator temperature transfer function

Equation (12) with the previously developed zero power reactor transfer function, given by equation (11), are incorporated in the block diagram below:

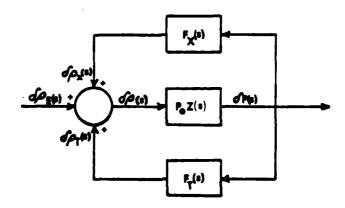


Figure 3. Block Diagram of the Reactor Transfer Function with Xenon-135 and Moderator Temperature Feedback Loops, and External Reactivity

The next step in the model development was to derive the Xenon-135 and moderator temperature transfer functions.

1. Xenon-135 Transfer Function

Xenon-135 is the most significant fission product poidon because of its enormous thermal neutron absorption cross section and relatively large fission yield. This isotope is not only produced directly from fission but also from the decay of other fission products as shown in Figure 4.

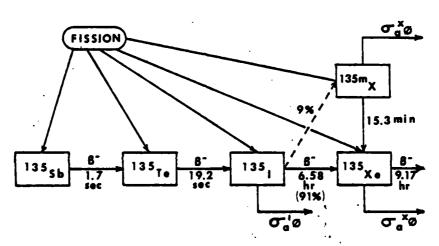


Figure 4. Xe-135 Decay Scheme

Since the β -decay of Iodine-135 (135 I) and Xenon-135 (135 Xe), with the largest half-lives, are the controlling steps in this decay scheme, it was simplified by making the following assumptions:

- 1) All 135 I is produced directly from fission (the production of Antimony-135 (135 Sb) and subsequent decay to 135 I is considered instantaneous).
- 2) The short lived metastable 135m Xe is ignored.

AL BELLEVICE TO STREET OF THE STREET

3) The removal of 135 I by neutron absorption is negligible for the neuton flux levels used in the model (10^{14} neutrons/cm²sec).

4) All the ^{135}I decays to ^{135}Xe .

With these assumptions the effective decay scheme is shown in Figure



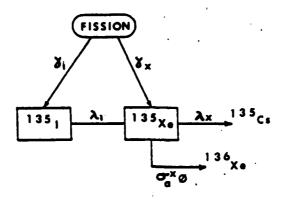


Figure 5. Simplified Xe-135 Decay Scheme

Using this decay scheme, the resulting rate equations for $^{135}\mathrm{I}$ and $^{135}\mathrm{Xe}$ are:

$$\frac{dI(t)}{dt} = \Im \sum_{i} \sum_{j} \emptyset(t) - \lambda_{i} I(t)$$
 (13)

$$\frac{\partial X(t)}{\partial t} = Y_X \sum_{f} \phi(t) + \lambda_I I(t) - \lambda_X X(t) - \sigma_{\bar{g}}^X \phi(t) X(t)$$
 (14)

where

 $I = {}^{135}I$ number density (atoms/cm 3)

 $X = {135}$ Xe number density (atoms/cm³)

 $\Sigma_f = Macroscopic fission cross section (cm⁻¹)$

 \emptyset = Integrated one group neutron flux (neutrons/cm²sec)

 χ_{τ} = Effective ¹³⁵I fission yield

 $\mathbf{X}_{\mathbf{Y}}$ = Effective ¹³⁵Xe fission yield

 $\lambda_{\tau} = ^{135}I$ decay constant (sec⁻¹)

 $\lambda_{x} = 135$ Xe decay constant (sec⁻¹)

 $\sigma_a^{\chi} = \frac{135}{\text{Xe microscopic thermal neutron absorption cross}}$

Again, using the first order perturbation technique, let

$$I(t) = I_{o} + \sigma I(t)$$

$$X(t) = X_{o} + \sigma X(t)$$

$$\emptyset(t) = \emptyset_{o} + \sigma \emptyset(t)$$
(15)

where the zero subscript denotes an initial steady state value and the delta prefix denotes a \underline{small} perturbation about this value.

Upon substituting equations (15) into equations (13) and (14), and taking the Laplace transform, the relationship between the perturbation in 135 Xe and Ø is derived (See Appendix A):

$$\frac{d^{2}X(5)}{d^{2}g(5)} = \frac{(3x\Sigma f - \sigma_{3}^{2}X_{o})s + \lambda_{1}(3z\Sigma f + 3x\Sigma f - \sigma_{3}^{2}X_{o})}{s^{2} + (\sigma_{3}^{2}g_{o} + \lambda_{K} + \lambda_{1})s + \lambda_{1}(\sigma_{3}^{2}g_{o} + \lambda_{K})} = G_{X}(s) \quad (16)$$

The change in reactivity caused by a small perturbation in \$135 Xe concentration is also derived in Appendix A:

$$\frac{\partial p_{X}(s)}{\partial x(s)} = -\frac{1}{X_o + \frac{G_b^F}{G_b^X}U} = \infty_X$$
 (17)

where

 X_0 = Equilibrium $\frac{135}{2}$ Xe number density before the perturbation (atoms/cm³)

U = Uranium-235 (235 U) number density (atoms/cm 3)

 σ_a^F = Microscopic thermal neutron absorption cross section of ^{235}U (cm²)

A perturbation in neuton flux (\mathscr{O} \emptyset) is directly proportional to a perturbation in reactor power (\mathscr{O} P) as shown by:

where E_f = Energy released per fission

 $\Sigma_{\rm f}$ = Macroscopic fission cross section of 235 U (cm⁻¹)

or

$$\frac{\partial \phi(s)}{\partial P(s)} = \frac{1}{E_f \Sigma_f} = K \tag{18}$$

The product of equations (17), (16), and (18) yields:

$$\frac{\partial / x(s)}{\partial / x(s)} \frac{\partial / x(s)}{\partial / g(s)} \frac{\partial / g(s)}{\partial / P(s)} = \frac{\partial / x(s)}{\partial / P(s)} = \alpha_{x} G_{x}(s) K = F_{x}(s)$$
(19)

Equation (19) is the transfer function relating reactor power and ¹³⁵Xe feedback reactivity. This equation is shown in block diagram form in Figure 6 where it has been incorporated with the previously derived zero power transfer function given by equation (11).

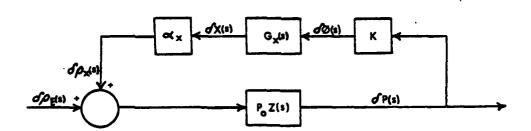


Figure 6. Block Diagram of the Reactor Transfer Function with Xenon-135 Feedback and External Reactivity

2. Moderator Temperature Feedback

Reactivity feedback from changes in moderator temperature occurs as a result of changes with moderator density. The density is also a

function of pressure, however, the pressure coefficient of reactivity is typically two orders of magnitude smaller than the temperature coefficient, and was therefore not considered in the model development. The moderator density affects the moderator number density and hence the macroscopic scattering cross section of the moderator.

The primary mechanism for the thermalization of the prompt and delayed neutrons is by elastic scattering interactions with the moderator nuclei. Hence, variations in the macroscopic scattering cross section will affect the rate at which neutrons become thermalized. Changes in the thermalization rate affect the fission rate, or reactor power level. Power level changes affect the moderator temperature, thus a feedback loop is created.

Because the moderator is also the coolant in the model, its temperature is not only a function of reactor power but also a function of the heat transfer process occurring in the heat exchanger, thus complicating the feedback loop.

Therefore, in order to develop this feedback mechanism analytically, it is first necessary to model the plant's heat transfer processes.

This thermal analysis is done in the following section.

Another temperature feedback mechanism is the broadening of the Uranium-238 resonance absorption cross section for neutrons with increasing temperature. Because a highly enriched core was assumed in the model, with little Uranium-238, this effect was not considered.

....

C. THERMAL ANALYSIS

1. Reactor Heat Transfer Function

A lumped parameter model was assumed. This simplification provided a set of ordinary differential equation which were sufficiently accurate for the simulated normal operating transients. In the lumped parameter model, heat transfer in the reactor was assumed to occur at a single point. Thus, spatial variations were neglected. The core was considered to be a homogenized mixture of the uranium alloy fuel, the fuel cladding, and other structural materials, with a constant thermal capacity.

The equation for the heat flow from the core was obtained from a basic heat balance. The heat generated from fission equals the heat required to change the temperature of the core materials plus the heat transferred to the coolant. On a per unit time basis, the heat balance is:

$$P(t) = C_{F} \frac{dT_{F}(t)}{dt} + h_{FM} \left[T_{F}(t) - T_{AV}(t) \right]$$
 (20)

where P(t) = Total Power generated in the core (Btu/sec)

 $T_c(t)$ = Average temperature of the core materials (°F)

 $T_{AV}(t)$ = Average reactor coolant temperature (°F)

 C_F = Total thermal capacity of the core materials (Btu/°F)

 h_{FM} = Total heat transfer coefficient (Btu/° F sec)

A similar heat transfer balance must also hold for the heat being transferred to the coolant and transported out of the core. This heat is the last term of equation (20) and is transferred to the coolant as:

$$h_{FM} \left[T_{F(e)} - T_{AV(e)} \right] = C_M \frac{dT_{AV(e)}}{dt} + \dot{m}_{m} C \left[T_{H_0(e)} - T_{C_1(e)} \right]$$
 (21)

where

C_M = Total thermal capacity of coolant in the core
 (Btu/°F)

 \hat{m}_m = Coolant mass flow rate (lbm/sec)

T_{H₀} = Average reactor coolant outlet temperature (°F)

T_C = Average reactor coolant inlet temperature (°F)

C = Specific heat of reactor coolant (Btu/lbm°F)

For simplification, the average reactor coolant temperature is assumed to be given by:

$$Tav(t) = \left[T_{ij}(t) + T_{ij}(t)\right]/2 \qquad (22)$$

Rearranging equation (20) yields:

$$T_{F}(t) + T_{1} \frac{dT_{F}(t)}{dt} = T_{AV}(t) + \frac{T_{1}}{C_{F}} P(t)$$
 (23)

where

$$T_1 = \frac{CF}{h_{FM}}$$

Substituting equation (22) into equation (21) to eliminate $T_{H_{\alpha}}$ yields:

Tav (e)
$$\left[1+2\frac{T_{2}}{T_{0}}\right]+T_{2}\frac{dT_{AV}(e)}{de}=T_{F}(e)+2\frac{T_{2}}{T_{0}}T_{C_{1}}(e)$$
 (24) where
$$T_{0}=\frac{C_{M}}{\dot{m}_{M}C}, \text{ and}$$

$$T_{Z}=\frac{C_{M}}{\dot{n}_{EM}}$$

Now let

$$P(t) = P_{0} + \sigma'P(t)$$

$$T_{F}(t) = T_{F_{0}} + \sigma'T_{F}(t)$$

$$T_{AV}(t) = T_{AV_{0}} + \sigma'T_{AV}(t)$$

$$T_{H_{0}}(t) = T_{Ho_{0}} + \sigma'T_{Ho}(t)$$

$$T_{C_{i}}(t) = T_{C_{i}} + \sigma'T_{C_{0}}(t)$$
(25)

where the zero subscript denotes a steady state value and the prefix a small perturbation about this value.

Substituting equations (25) into equations (22), (23), and (24), eliminating the average temperatures $T_F(t)$ and $T_{AV}(t)$, and taking Laplace transforms, the following relationship between reactor coolant inlet and outlet temperature perturbations is derived (see Appendix B):

$$\frac{\partial^{3}T_{H_{0}}(s)}{\left\{\frac{\mathcal{T}_{1}}{2}s^{2} + \left[\frac{\mathcal{T}_{0}}{\mathcal{T}_{2}}\left(\frac{\mathcal{T}_{1}}{\mathcal{T}_{2}} + 1\right) - \mathcal{T}_{1}\right]s - 1\right\}\mathcal{O}^{3}T_{C_{1}}(s) + \mathcal{O}^{3}P(s)}{\left\{\frac{\mathcal{T}_{1}}{2}s^{2} + \left[\frac{\mathcal{T}_{0}}{\mathcal{T}_{2}}\left(\frac{\mathcal{T}_{1}}{\mathcal{T}_{2}} + 1\right) + \mathcal{T}_{1}\right]s + 1\right\}}$$
(26)

where

$$\lambda = \frac{\vec{1}_1 \vec{1}_0}{\vec{C}_F \vec{1}_2} = \frac{1}{\dot{m}_H C}$$

This equation is the reactor heat transfer function for the reactor coolant outlet temperature as a function of the reactor coolant inlet temperature and reactor power.

A block diagram representation of this transfer function is shown in Figure 7.

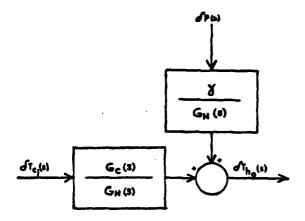


Figure 7. Block Diagram of the Reactor Heat Transfer Function

In Figure 7

$$G_{\mathbb{C}}(S) = \left\{ \frac{7.7}{2} S^2 + \left[\frac{7.7}{7.2} \left(\frac{7.7}{7.2} + 1 \right) - \frac{7.7}{4} \right] S + 1 \right\}$$

$$G_{H}(S) = \frac{1}{\left\{\frac{7/72}{2}s^{2} + \left[\frac{7e}{72}\left(\frac{7/2}{72} + 1\right) + \frac{7}{4}\right]S + 1\right\}}$$

2. Heat Exchanger Transfer Function

As in the derivation of the reactor heat transfer function, a lumped parameter model was assumed for the heat exchanger. Two points of energy storage were assumed, the primary coolant water and the water on the secondary side of the heat exchanger. The thermal capacity of the heat exchanger metal was included with that for the secondary water, since the thermal resistance on the primary side is predominant. A further assumption was made that the time spent by the primary coolant

while it passed through the heat exchanger was negligible in comparison with the time spent in the primary piping.

The state of the steam produced on the secondary side of the heat exchanger was assumed to always be dry and saturated and that the secondary water is always at the saturation temperature for the existing pressure. These assumptions were justified because moisture separators and recirculation can provide high quality steam and preheating of the feedwater.

With these assumptions, the following equations were obtained from a heat balance per unit time:

$$\dot{m}_{M} \subset \left[T_{H_{1}}(t) - T_{C_{0}}(t) \right] = C_{M} \frac{dT_{AV_{0}}(t)}{dt} + h_{TM} \left[T_{AV_{0}}(t) - T_{0}(t) \right]$$
 (27)

$$h_{TM} \left[T_{AV_B}(t) - T_S(t) \right] = C_S \frac{dT_S(t)}{dt} + P_c(t)$$
 (28)

where T_{u;}(t) = Heat exchanger coolant inlet temperature (°F)

 $T_{Co}(t)$ = Heat exchanger coolant outlet temperature (°F)

 $T_{AV_R}(t)$ = Heat exchanger average coolant temperature (°F)

 $T_c(t)$ = Saturated steam temperature (°F)

h_{TM} = Total heat transfer coefficient (BTU/°F.sec)

C_M = Total thermal capacity of coolant in the heat exchanger (BTU/°F)

C_S = Total thermal capacity of heat exchanger metal
 and secondary water and steam (Btu/°F)

 $P_i(t)$ = Power delivered by the heat exchanger (BTU/sec)

 m_m = Coolant mass flow rate (1bm/sec)

The total heat transfer coefficient, a function of the primary coolant flow rate (a constant in the model) and the heat transfer characteristics of the heat exchanger, was assumed to be constant. Also, the time delay in transferring heat across the heat exchanger tubes was neglected. This changed the shape of the initial thermal transient but had little effect on the basic dynamics of the secondary loop.

Again, as a simplification, the average temperature of the coolant in the heat exchanger was assumed to be given by

$${}^{T}AV_{B} = \frac{1}{2} (T_{H_{1}} + T_{C_{0}})$$
 (29)

The power delivered by the steam generator is proportional to the product of the steam flow rate and the difference in enthalpy between the steam and feedwater

$$h_{TM} \left[T_{AVB}(t) - T_{S}(t) \right] = \dot{m}_{S} \left(H_{S} - H_{FW} \right) + C_{S} \frac{dT_{S}(t)}{dt}$$

where

 \dot{m}_s = Steam flow rate (1bm/sec)

 H_S = Saturated steam enthalpy (Btu/lbm)

 H_F = Feedwater enthalpy (Btu/lbm)

This equation assumes the steam and feedwater flow rates are always equal and thus neglects any instabilities in the secondary steam.

As discussed in Reference 4, the enthalpy of saturated steam is nearly constant over a wide range of pressure, varying from 1198 Btu/1bm to 1204 Btu/1bm over a pressure range from 200 to 800 psig. The feed-water enthalpy, which depends on condenser pressure, is usually between 50 and 100 Btu/1bm. Thus the enthalpy difference may be regarded as a

constant, and the power delivered by the heat exchanger is directly proportional to the steam flow rate.

The impedance to steam flow caused by the turbine is nearly independent of turbine speed. If constant backpressure is assumed, the steam flow rate is directly proportional to the throttle opening at a given pressure. Thus,

where

$$P_S$$
 = Saturated steam pressure (lbf/in²)

A = Proportionality factor which is a function of the throttle setting lbm in²
lbf sec

The numerous assumptions made in the heat exchanger model development limit the accuracy of the resulting equations. However, the errors involved in these assumptions are usually less than the amount of uncertainty in the engineering valves of the coefficients used in the equations.

Recalling the previous assumption that

$$\frac{dT_{AV_B}(t)}{-dt} = 0 , and$$

Substituting equation (29) into equation (27), yields:

$$\dot{m}_{\mathsf{M}} C \left[T_{\mathsf{H}_{1}}(\mathsf{e}) - T_{\mathsf{c}_{0}}(\mathsf{e}) \right] = h_{\mathsf{TM}} \left\{ \frac{1}{2} \left[T_{\mathsf{H}_{1}}(\mathsf{e}) + T_{\mathsf{c}_{0}}(\mathsf{e}) \right] - T_{\mathsf{S}}(\mathsf{e}) \right\}$$
(30)

Solving equation (30) for T_{C_0} , gives:

$$Tc_0(e) = \frac{2T_3(e) - T_{H_1}(1-K_1)}{(1+K_1)}$$
 (31)

where
$$K_1 = \frac{2 \dot{m}_M C}{h_M}$$

Substituting equation (29) into equation (28) gives:

$$h_{TM}\left\{\frac{1}{2}\left[T_{H_1}(t)+TC_0(t)\right]-T_S(t)\right\}=C_S\frac{dT_S(t)}{dt}+P_L(t)$$

Rearranging this expression,

$$\frac{Cs}{h_{TM}} \frac{dT_S(t)}{dt} + T_S(t) = \frac{1}{2} \left[T_{H_1}(t) + T_{C_0}(t) \right] + \frac{1}{h_{TM}} P_L(t)$$
 (32)

Let
$$T_{Hi}(t) = T_{H_{i0}} + dT_{H_{i}}(t)$$

 $T_{C_{0}}(t) = T_{C_{0_{0}}} + dT_{S}(t)$ (33)
 $T_{S}(t) = T_{X_{0}} + dT_{S}(t)$
 $P_{L}(t) = P_{L_{0}} + dP_{L}(t)$

where the lowest zero subscript denotes a steady state value and the delta prefix a small perturbation about this value.

By substituting equations (33) into equations (31) and (32), and taking the Laplace transform, the following set of equations is derived (see Appendix C):

$$\int T_{S}(s) = \frac{\frac{1}{2} \left[\int T_{C_{O}}(s) + \int T_{H_{1}}(s) \right] - \frac{1}{h_{TM}} \int P_{L}(s)}{T_{S} + 1} \tag{34}$$

$$\mathcal{J}T_{C_0}(s) = \frac{2\mathcal{J}T_S(s) - \mathcal{J}T_{H_1}(s)[1-K_1]}{1+K_1}$$
where
$$\mathcal{T}_S = \frac{C_S}{h_{TM}}$$
(35)

These equations are represented in a block diagram in Figure \$.

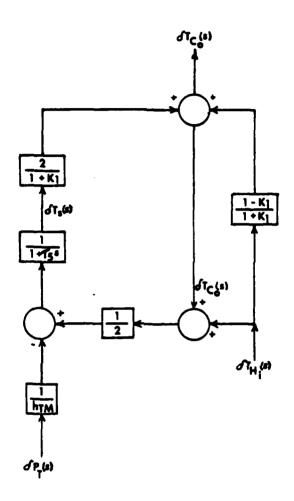


Figure 8. Block Diagram of the Heat Exchanger Transfer Function

With $\mathcal{S}T_{H_1}$ and $\mathcal{S}P_L$ as inputs, equations (34) and (35) form an algebraic loop in $\mathcal{S}T_{C_0}$ and $\mathcal{S}T_{S}$. These two equations were solved with the use of an inherent functional routine available in the CSMP-III language which was used to formulate the model.

3. Primary Piping Transfer Functions

While circulating through the primary loop, the coolant undergoes mixing and transport delay effects. For example, a transient in the coolant temperature at the heat exchanger outlet does not appear at the reactor inlet until sometime later. Where there are volume or flow direction changes, as in the reactor coolant inlet plenum, mixing occurs causing a smearing of a temperature transient. Both of these effects were approximated in the model by combinations of two types of time delays: a pure transport delay and a simple mixing delay.

Assuming no mixing in the primary piping and no heat loss with perfectly insulated pipes, pure transport delays are encountered in the piping runs between the reactor and heat exchanger. With these assumptions, the temperatures involved in the transfer of heat between the reactor and heat exchanger can be given as

$$T_{H_1}(t) = T_{H_0}(t - T_3)$$

where the inlet coolant temperature to the heat exchanger $T_{H_{\hat{i}}}$ has the same form as the outlet temperature of the reactor $T_{H_{\hat{o}}}$ after a fixed transport delay $T_{\hat{3}}$. This transport delay can be approximated by the following differential equation which is derived in Appendix D.

$$T_{H_{i}} + T'_{3} = T_{H_{0}}$$
 (36)

Similarly, the transport delay from the heat exchanger outlet to the reactor inlet plenum is:

$$T_{C_{ip}}(t) + T_4 = \frac{dT_{C_{ip}}(t)}{dt} = T_{C_0}(t)$$
 (37)

where $T_{C_{ip}}(t)$ = Reactor inlet plenum coolant temperature (°F) $T_{C_{o}}(t)$ = Heat exchanger outlet coolant temperature (°F)

This can be approximated by

$$T_{C_{ip}}(t) + I_4 \frac{dT_{C_{ip}}(t)}{dt} = T_{C_0}(t)$$
 (37)

As in the previous development, let

$$T_{H_{0}}(t) = T_{H_{00}} + \sigma T_{H_{0}}(t)$$

$$T_{H_{1p}}(t) = T_{H_{1p0}} + \sigma T_{H_{1p}}(t)$$

$$T_{C_{0}}(t) = T_{C_{0}} + \sigma T_{C_{0}}(t)$$

$$T_{C_{1p}}(t) = T_{C_{1p0}} + \sigma T_{C_{1p}}(t)$$
(38)

where again the lowest zero subscript denotes a steady state value and the delta prefix a small perturbation about this value.

After substituting equations (38) into equations (36) and (37), taking the Laplace transform, the following transfer functions are derived (see in Appendix D).

$$\frac{d^{7}T_{H_{0}}(s)}{d^{7}T_{H_{0}}(s)} = \frac{1}{1+7_{3}s}$$
 (39)

$$\frac{\partial^{T}C_{4p}(s)}{\partial^{T}C_{0}(s)} = \frac{1}{1+\tilde{1}45}$$
 (40)

Combined mixing and transport effects are encountered at the reactor and heat exchanger inlet and outlet coolant plenums. For simplification, only the mixing effects at the inlet plenums were considered in the model. Assuming perfect mixing and after performing a heat balance on the plenum concerned, the combined mixing and transport effects are expressed by differential equations developed in Appendix D of the form

where

 $T_i(t) = Plenum inlet temperature (°F)$

 $T_{O}(t) = Plenum outlet temperature (°F)$

1 = Mixing delay

As before, by using a <u>small</u> perturbation technique, and taking the Laplace transform, the following transfer function is derived (see Appendix D):

$$\frac{\sqrt{T_{H_0}(s)}}{\sqrt{T_{H_1}(s)}} = \frac{1}{1+7s} \tag{41}$$

Block diagram representations of the coolant piping transport and mixing transfer functions is shown in Figure 9.

D. OVERALL PLANT BLOCK DIAGRAM

The transfer functions developed previously for the zero power point reactor, 135 Xe feedback, reactor and heat exchanger heat transfer, and the primary piping were interconnected resulting in the overall model block diagram shown in Figure 10. The moderator temperature feedback

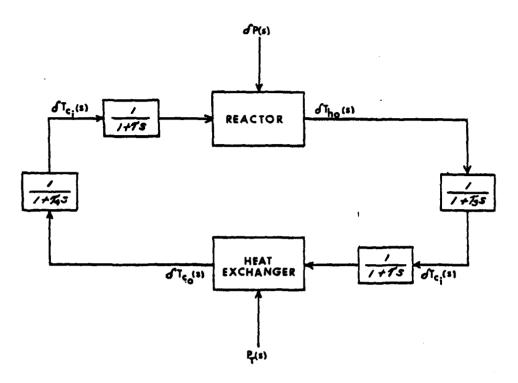


Figure 9. Block Diagram of the Transport and Mixing Delay Functions

loop is seen to consist of the heat transfer process and primary piping transfer functions which yield $\int_{C_i}^{T_{C_i}} T_{C_i}$ and $\int_{O}^{T_{H_o}}$. These temperatures are summed, then halved, yielding T_{AV} , which is then multiplied by the negative temperature coefficient \propto T, generating the moderator temperature feedback reactivity ρ_T .

E. MODEL CONSTANTS

The following plant parameters from the Shippingport Atomic Power
Station were obtained from Reference 6 and used in the model development.

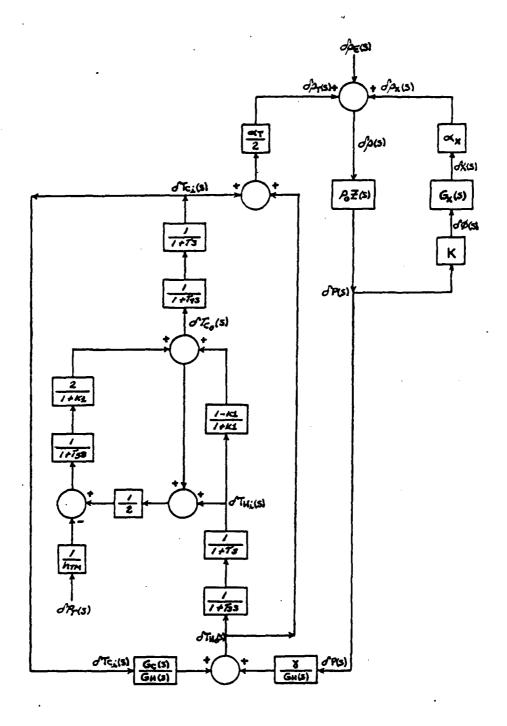


Figure 10. Overall Plant Block Diagram

1. General Parameters

٠.	den	era rarameters		
	a.	Reactor thermal power	231 MW	
	b.	Reactor coolant system pressure	2000 psia	
	c.	Reactor coolant average temperature	523° F	
	d.	Steam pressure at full load	600 psia	
2.	Rea	ctor Coolant System		
	a.	Reactor coolant flow rate	6280 1bm/sec	
	b.	Reactor coolant outlet temperature	538° F	
	c.	Reactor coolant inlet temperature	508° F	
	d.	Coolant volume in core	103 ft ³	
3.	Rea	ctor Core		
	a.	Configuration	Right cylinder	
	b.	Size	6.8 ft. dia X 6 ft. high	
	с.	Fuel load U ²³⁵ (seed)	75 kg	
	d.	Composition (seed)		
		1) Water	43.5 v/o	
		2) Fuel alloy	30.0 v/o	
		3) Zircalloy	34.1 v/o	
	e.	Control Rods		
		1) Total rod worth	0.256 k	
		2) Scram time	0.35 sec time delay	
			1.0 sec rod drop	
4.	Nuc	Nuclear Data		
	a.	Thermal neutron flux	$2 \times 10^{14} \text{ n/cm}^2 \text{ sec}$	
	b.	Prompt neutron lifetime	5.6 X 10 ⁻⁵ sec	

c. Effective delayed neutron fraction

0.0077

d. Temperature coefficient of reactivity -3.1 X 10⁻⁴ d°k/°F

5. Reactor Protection Setpoints

a. Scram

1) High reactor power

138%

2) High reactor coolant outlet temperature

550°F

b. Cutback

1) High reactor power

114%

2) High Startup rate

1.74 Decades per min.

IV. RESULTS

A. INTERACTIVE PROGRAM

Figures 10-16 are examples of the interactive program output. As shown in Figure 10, the user is first given a description of the purpose of the overall program and general instructions for input entry. As shown in Figure 11, the user is then prompted to enter an initial steady state power level, after which the corresponding major plant parameters are displayed. The user is then prompted to choose the type of simulation, either plant transient at power, or post-shutdown Xenon-135 behavior. The user must be at the Tektronix 4012 console in the computer center if the latter is chosen.

If the post-shutdown Xenon-135 behavior is chosen to be examined, graphs such as those shown in Figure 13 are generated and displayed on the Tektronix's screen. In addition to showing the time response of the Xenon, the time of its peak and associated maximum reactivity are displayed.

If the plant transient simulation is chosen, the user is then prompted to choose the mechanism for initiating the transient, either control rod movement or turbine load change. In Figure 14, the user has chosen the former. The program then prompts the user to enter the direction and time length of the control rod movement. In Figure 15, the turbine load change was chosen and the program request the final turbine load. In either case, a summary of the transient inputs is displayed with a notice that the interactive portion is complete.

EXECUTION BEGINS...

OF A 231 MWT (48 MWE), HIGHLY ENRICHED PRESSURIZED WATER REACTOR (PWR), FOWER PLANT TO CHANGES IN REACTIVITY INITIATED BY CHANGES IN BANK CONTROL ROD POSITION OR TURBINE LOAD, THE REACTIVITY FEEDBACK EFFECTS RESULTING FROM CHANGES IN MODERATOR TEMPERATURE AND XENON-135 CONCENTRATION ARE INCORPORATED INTO THE MODEL. THIS IS AN INTERACTIVE PROGRAM WHICH SIMULATES THE THERMAL RESPONSE

FLEASE ANSWER THE FOLLOWING STATEMENTS BY ENTERING THE NUMBER (INCLUDING THE DECIMAL FOINT, BUT NOT THE PERCENT SYMBOL) WHICH CORRESPONDS TO THE DESIRED CHOICE OR VALUE WITHIN THE GIVEN RANGE.

DO YOU WANT A BRIEF DESCRIPTION OF THE FLANT? 1. YES

>2.

Figure 11. Interactive Program, Introduction and Instructions

ENTER AN INITIAL STEADY STATE POWER LEVEL BETWEEN 10. AND 100. PERCENT. NOTE: A MINIMUM OF 10% POWER IS REQUIRED TO OPERATE THE REACTOR COOLANT PUMPS AND OTHER PLANT AUXILIARIES

>50. INITIAL STEADY STATE PARANETERS: REACTOR POWER= 50,00% 115,50 MWE

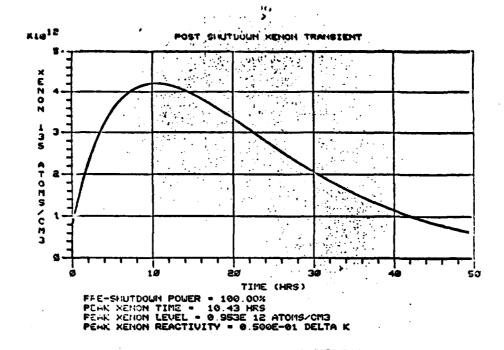
REACTOR POWER= 50.002 115.50 MWE
THERMAL NEUTRON FLUX= 0.10E 15 NEUTRONS/CM2 SEC
XENON-135 CONCENTRATION= 0.76E 12 ATOMS/CM3
FUEL TEMPERATURE=711.50 F
AVERAGE REACTOR CODLANT MODERATOR) TEMPERATURE=523.00 F
REACTOR COOLANT DUTLET TEMPERATURE=530.50 F
REACTOR COOLANT INLET TEMPERATURE=515.50 F
SATURATED STEAM TEMPERATURE=504.54 F
SATURATED STEAM FRESSURE=704.07 PSIA

TURBINE LOAD= 50.00% 34.00 MWE

ENTER THE DESIRED SIMULATION: 1. MODERATOR TEMPERATURE AND XENON FEEDBACK AT POWER

2. POST SHUTDOWN XENON BEHAVIOR

Interactive Program, Initial Power Level and Simulation Choice Inputs Figure 12.



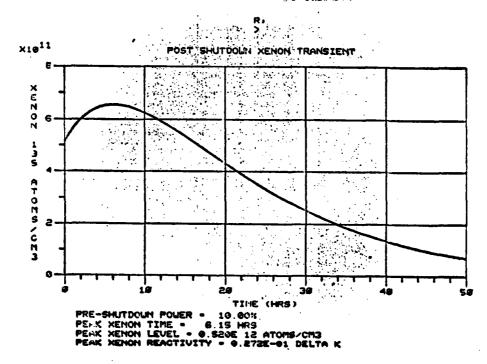


Figure 13. Post-shutdown Xenon-135 Behavior

ENTER THE INITIAL REACTIVITY CHANGE MECHANISM:

1. BANK CONTROL ROD CHANGE

2. TURBINE LOAD CHANGE

<u>`</u>

THE TIME LENGTH (SEC) OF BANK CONTROL ROD MOVEMENT (TSHIM)
BANK CONTROL ROD REACTIVITY INSERTION RATE IS 1.25E-4 DELTA K/SEC,
DO NOT EXCEED A TOTAL REACTIVITY INSERTION OF 35.0E-4 DELTA K (I.E. A 28 SEC SHIM) ENTER NOTE: >10.

ENTER THE DIRECTION OF BANK CONTROL ROD SHIH: 1. IN (NEGATIVE REACTIVITY INSERTION) 2. OUT (POSITIVE REACTIVITY INSERTION)

SHIM TIME = 10.00 SEC SHIM DIRECTION = 1.00 (1=*OUT*, -1=*IN*) FINAL TURBINE LOAD = 50.00% INITIAL REACTIVITY CHANGE FARAMETERS: REACTOR POWER = 50.00 %

INPUTS HAVE BEEN ENTERED INTO THE SIMULATION FROGRAM WHICH HAS BEEN SENT TO THE BATCH SYSTEM FOR EXECUTION RJ

Interactive Program, Control Rod Movement Inputs Figure 14. ENTER THE INITIAL REACTIVITY CHANGE MECHANISH:

1. BANK CONTROL ROD CHANGE

2. TURBINE LOAD CHANGE

>2.

ENTER FINAL TURBINE LOAD BETWEEN 10. AND 100. PERCENT NOTE: LOAD VARIATION WILL OCCUR AT A FIXED RATE OF 0.5%/SEC >75.

INITIAL REACTIVITY CHANGE FARAMETERS;

REACTOR FOWER = 50.00 %

SHIM TIME = 0.0 SEC

SHIM DIRECTION = -0.25 (1="OUT", -1="IN")

FINAL TURBINE LOAD = 75.00%

INPUTS HAVE BEEN ENTERED INTO THE SIMULATION PROGRAM WHICH HAS BEEN SENT TO THE BATCH SYSTEM FOR EXECUTION R\$

Interactive Program, Turbine Load Change Inputs Figure 15.

Figure 16 is an example of the interactive program's logic routines which will only permit inputs within the requested ranges to be accepted for the simulation.

B. SIMULATION PROGRAM

The program generates Versatec plots showing the plant's power, reactivity, and temperature transient responses for the user's inputs. Figures 17-20 are composites of these plots.

In Figure 17, the transient was initiated by a simulated 20 to 40 percent ramp change in turbine load at 1/2 percent per second. In Figure 18, the transient was initiated by a simulated 60 to 40 percent change in turbine load in the same manner. Both of these figures show the characteristic power demand following and inherent stability response typical of a PWR plant with a constant average temperature program and negative temperature coefficient.

The inherent stability feature is further displayed in Figures 19 and 20. In Figure 19, the transient is initiated by a simulated 10 second inward movement of the control rods at a reactivity insertion rate of 1.25 x 10⁻⁴ Jk per second from an initial 50 percent power level. In Figure 20, the transient response to an outward control rod movement with the same parameters is shown. As seen in both cases, the reactor power stabilizes about the initial turbine load and the average reactor coolant temperature stabilizes at a new level to compensate, via the negative temperature coefficient, for the reactivity inserted by the control rods.

ENTER AN INITIAL STEADY STATE POWER LEVEL BETWEEN 10. AND 100. PERCENT. NOTE! A MINIMUM OF 10% POWER IS REQUIRED TO OPERATE THE REACTOR COOLANT FUMPS AND OTHER PLANT AUXILIARIES

>150.

INCORRECT DATA ENTERED

A NINIMUM OF 10% POWER IS REGUIRED TO OPERATE THE REACTOR COOLANT PUMPS AND OTHER PLANT AUXILIARIES ENTER AN INITIAL STEADY STATE POWER LEVEL BETWEEN 10. AND 100. PERCENT. NOTE:

× ×

INCORRECT DATA ENTERED

ENTER AN INITIAL STEADY STATE POWER LEVEL BETWEEN 10. AND 100. PERCENT. NOTE: A HINIMUM OF 10% POWER IS REQUIRED TO OPERATE THE REACTOR COOLANT PUMPS AND OTHER PLANT AUXILIARIES

Figure 16. Interactive Program, Logic Routine Example

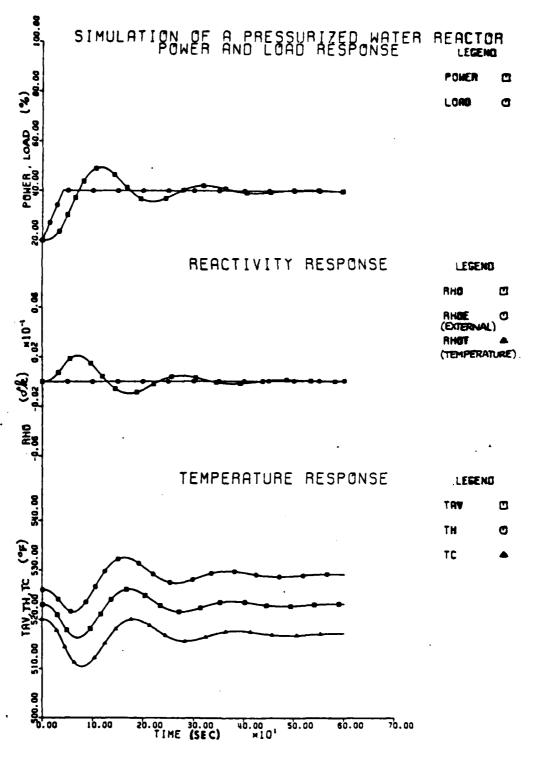


Figure 17. Power, Reactivity, and Temperature Response to a 20 to 40% Turbine Load Change

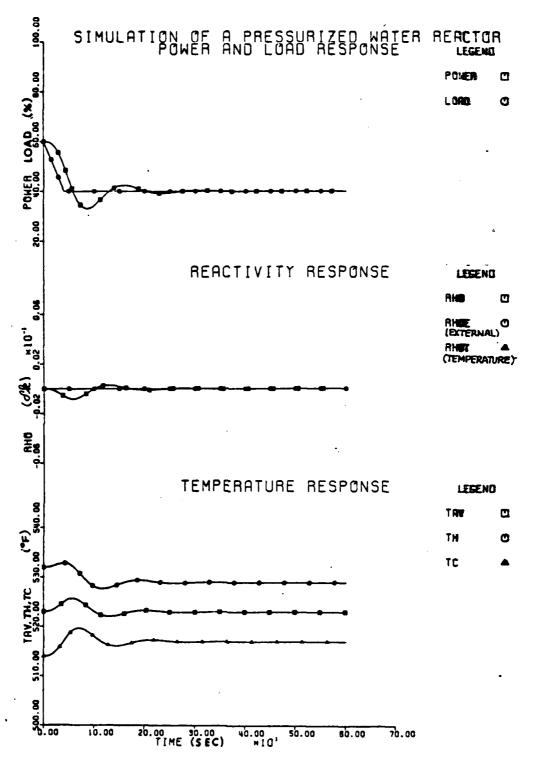


Figure 18. Power, Reactivity, and Temperature Response to a 60 to 40% Turbine Load Change

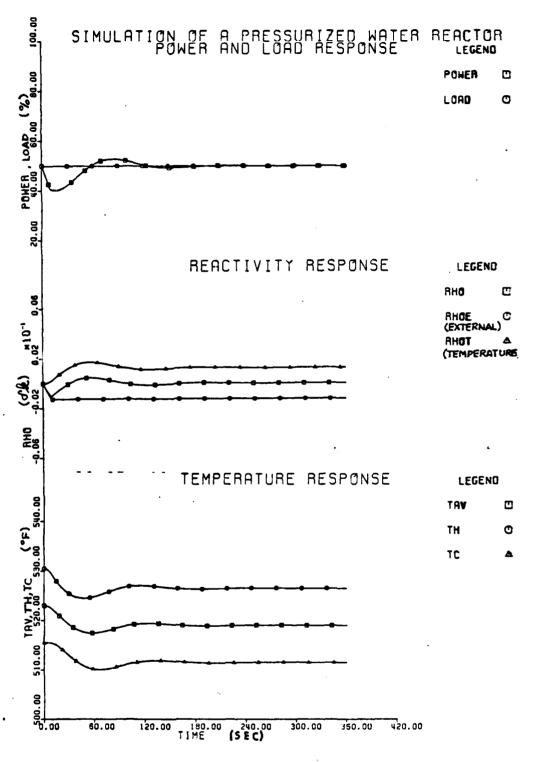


Figure 19. Power, Reactivity, and Temperature Response to a 10 second Inward Control Rod Movement

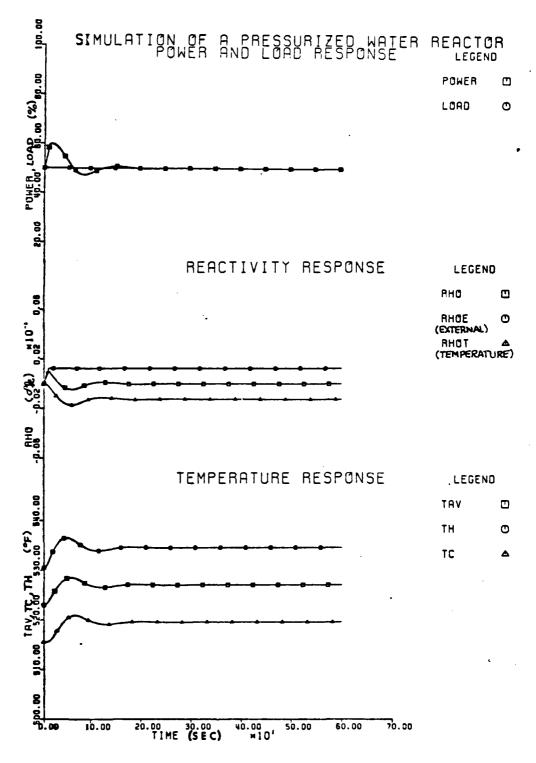


Figure 20. Power, Reactivity, and Temperature Response to a 10 second Outward Control Rod Movement

V. CONCLUSIONS AND RECOMMENDATIONS

The model responds to normal operating transients with the characteristics of a PWR plant. The interactive program provides the user with the facility to initiate the simulation or examine a real time display of post-shutdown Xenon-135 buildup and decay.

As previously discussed, the purpose of this work was to develop a learning device for nuclear engineering students at the Naval Postgraduate School. The result was a relatively simple model of a PWR power plant and a simulation program that suffers from a long execution time.

The model's simplicity resulted mainly from the consideration of an effective single group of delayed neutron precursors in the reactor kinetic equations and the use of lumped parameter analysis in the plant's heat transfer processes. The model's sophistication could be increased by:

- 1. Expanding the reactor kinetic equations to consider six delayed neutron precursor groups.
- 2. Developing a multi-section heat transfer model for the reactor and the heat exchanger.
- 3. Incorporating variable reactor coolant mass flow rate and heat transfer coefficients.

These additional features will also increase the complexity of the simulation program, thus aggravating the long run time problem. This concern might be eased by taking a different approach to the model's formulation than the transfer function method. While this method was readily coded using the CSMP-III language, physical insight to the plant dynamic processes was lost when the Laplace transform and subsequent grouping of constant terms was performed. The use of state variable

theory might not only allow this physical appreciation to be retained but also result in shorter program run times.

Regardless of the formulation method, a real time response to the simulation can probably only be achieved by using an analog computer. The wide range of time constants associated with the equations (prompt neutron life-times on the order of 10⁻⁵ seconds to Xenon decay half lives on the order of hours) require a small numerical integration time interval over a large time period. The result is a prohibitively long run time on a digital computer for an interactive program.

It is hoped that refinement of the model will be continued. While the existing model does reflect the general characteristics of a PWR plant to normal operating transients, it does exhibit relatively large power over/under shoots in response to turbine load changes as seen in Figures 17 and 18.

APPENDIX A

DEVELOPMENT OF THE XENON FEEDBACK TRANSFER FUNCTION COMPONENTS

1. Derivation of $G_{x}(S)$

The previously developed rate equations for $^{135}{\rm I}$ and $^{135}{\rm Xe}$ from the effective $^{135}{\rm Xe}$ decay scheme shown in Figure 5 are:

$$\frac{dt}{dI} = \chi_{I} \sum l \varphi(t) - \gamma_{I} I(t)$$
 (V 1)

$$\frac{dX}{dt} = \lambda x \sum f \phi(\epsilon) + y z I(\epsilon) - y^x \chi(\epsilon) - a_x \chi(\epsilon) \chi(\epsilon) \qquad (4.5)$$

where $I(t) = {}^{135}I$ number density (atoms/cm³)

 $X(t) = ^{135}Xe$ number density (atoms/cm³)

 Σ f = Macroscopic fission cross section of ^{235}U (cm⁻¹)

Ø = Average integrated one group flux (neutrons/cm²sec)

 $\delta_{\rm I}$ = Effective ¹³⁵I fission yield

 χ_{X} = Effective ¹³⁵ Xe fission yield

 $\lambda_{\rm I} = ^{135}{\rm I} \, {\rm decay \, constant \, (sec^{-1})}$

 $\lambda_{X} = ^{135}Xe \text{ decay constant (sec}^{-1})$

 $G_{\lambda}^{\chi} = {}^{135}Xe$ thermal neutron absorption cross section (cm²)

Let
$$I(t) = I_0 + \partial^{t}I(t)$$

$$X(t) = X_0 + \partial^{t}X(t)$$

$$\emptyset(t) = \emptyset_0 + \partial^{t}\emptyset(t)$$
(A 3)

where the zero subscript denotes an initial steady state value and the delta prefix denotes a small perturbation about this value. With the reactor at an initial steady state with a flux level \emptyset_0 , the equilibrium values of ^{135}I and ^{135}Xe are found by setting their time dependence in equations (A 1) and (A 2) equal to zero. Thus,

$$I_o = \frac{\lambda_z \sum f \phi_o}{\lambda_z}$$
 (A 4)

$$X_o = \frac{y_x \sum_f \phi_o + \lambda_I I_o}{\lambda_x + \sigma_a^x \phi_o}$$
 (A 5)

Substituting equation (A 4) into equation (A 5)

$$X^{\circ} = \frac{y^{x} + \alpha y^{2}}{(x^{x} + x^{2}) \sum_{i} Q^{\circ}}$$
 (A 6)

$$\frac{d}{dt} dT = \lambda_I \sum_i d \delta_i - \lambda_I d I \qquad (A 7)$$

$$\frac{df}{dx} q_X = y I q_I + (\chi_X \Sigma t - c_X \chi^0) q_X - (\chi^X + c_X q^0) q_X \qquad (48)$$

Taking the Laplace transform of equations (A 7) and (A 8) and solving for $\sigma I(S)$ and $\sigma X(S)$

$$\sigma^{T}(s) = \frac{s + \lambda r}{s + \lambda r}$$
 (A 9)

$$d^{\prime}X(s) = \frac{\lambda r d^{\prime}I(s) + (\lambda x \Sigma t - \alpha_{x}^{\prime}X^{\circ}) d^{\prime}\phi(s)}{(\lambda r)^{\prime}}$$
(A10)

Substituting equation (A 9) into equation (A10)

$$d^{2}X(s) = \frac{\lambda_{2} \lambda_{3} \sum_{r} d^{2}g(s) + (s + \lambda_{2})(\lambda_{2} \sum_{r} - \alpha_{3} \lambda_{4}) d^{2}g(s)}{(s + \lambda_{2})(s + \lambda_{3})(\lambda_{3} \sum_{r} - \alpha_{3} \lambda_{4}) d^{2}g(s)}$$

Expanding, collecting terms, and solving for

$$\frac{\partial^2 X(S)}{\partial^2 G(S)} = \frac{\left(3 \times Z_f - \sigma_a^2 X_o\right) S + \lambda_I \left(3 \cdot Z_f + 3 \times Z_f - \sigma_a^2 X_o\right)}{S^2 + \left(\sigma_a^2 M_o + \lambda_X + \lambda_I\right) S + \lambda_I \left(\sigma_a^2 M_o + \lambda_X\right)} = G_X(S)$$

This is equation (16) on page 23.

Derivation of

The effective core neutron multiplication factor k is given by the familiar "six factor formula" found in the literature

$$k = N \in \mathcal{P}_{FNL} P_{TNL}$$
 (A 11)

where \mathcal{I} = Thermal fission factor of the fule

€ = Fast fission factor of the fuel

p = Resonance escape probability

f = Thermal utilization factor in the core

P_{FNI} = Fast neutron non-leakage probability

= Thermal neutron non-leakage probability

The core reactivity is defined as

$$\rho = \frac{k-1}{k} \tag{A 12}$$

Consider a reactor at an initial critical steady state condition with an equilibrium 135 Xe concentration Xo. By the definition of criticality

$$ko = 1$$
 (A 13) $\rho = 0$

In this condition

$$f_o = \frac{\Sigma_a^F}{\Sigma_{a_o}^{\text{obse}}} \approx \frac{\Sigma_a^F}{\Sigma_a^F + \Sigma_a^{X_o}}$$
(A 14)

where

∑_a = Fuel macroscopic thermal neutron absorption cross section

zore = Initial core macroscopic thermal neutron absorption cross section

 $\sum_{a}^{X_0}$ = Initial ¹³⁵Xe macroscopic thermal neutron absorption cross section

and the macroscopic thermal neutron absorption cross sections of the moderator and other core materials has been neglected, a reasonable approximation in a highly enriched core.

Now impose a small perturbation in the 135 Xe concentration on the core. Neglecting the effect of the 135 Xe perturbation on neutron leakage, a reasonable assumption for a large reactor,

$$X = Xo + d^{2}X$$

$$k = ko + d^{2}k$$

$$f = fo + d^{2}f$$

$$\rho = \rho_{o} + d^{2}\rho = d\rho \quad \text{since } \rho_{o} = 0$$

The remainder of the terms in equation (A 11) are unchanged, thus

$$dk = df$$

and

$$\frac{k}{ko} = \frac{f}{fo} = k$$
 since $ko = 1$

where

$$f = \frac{\sum_{a}^{F}}{\sum_{a}^{core}} = \frac{\sum_{a}^{F}}{\sum_{a}^{core} + \sigma_{a}^{2} \sigma_{x}^{0} \chi}$$
 (A 15)

from equation (A 12) and (A 13)

substituting equations (A 14) and (A 15) into equation (A 16)

$$\frac{\partial^{2} x}{\partial x} = 1 - \frac{\sum_{a_{o}}^{ore} + \sigma a^{x} \partial^{1} x}{\sum_{a_{o}}^{ore}} = -\frac{\sigma a^{x} \partial^{1} x}{\sum_{a_{o}}^{ore}} = -\frac{\sigma a^{x} \partial^{1} x}{\sigma a^{x} x_{o} + \sum_{a}^{F}}$$
or
$$\frac{\partial^{2} x}{\partial^{1} x} = -\frac{1}{x_{o} + \frac{\sigma a^{u}}{\sigma a^{x}} U} = \infty_{X}$$

This is equation (17) on page 23. Since fuel depletion effects are not considered during the short simulated transients, the fuel number density U is assumed constant and \propto_X is invariant in the time and the Laplace domains.

APPENDIX B

DEVELOPMENT OF THE REACTOR HEAT TRANSFER FUNCTIONS

The previously developed equations describing heat transfer from the reactor to the coolant are

$$T_F(t) + I_1 \frac{dT_F(t)}{dt} = T_{AV}(t) + \frac{T_1}{C_F} P(t)$$
 (B 1)

$$T_{AV}(t)\left[1+2\frac{T_2}{T_0}\right]+12\frac{dT_{AV}(t)}{dt}=T_F(t)+2\frac{12}{T_0}T_{C_1}(t)$$
 (B 2)

$$T_{AV}(t) = \frac{1}{2} \left[T_{Ho}(t) + T_{C_1}(t) \right]$$
 (B 3)

 $T_F(t)$ = Average core material temperature (°F) where

 $T_{AV}(t)$ = Average reactor coolant temperature (°F)

 $T_{Ho}(t)$ = Reactor coolant outlet temperature (°F)

 $T_{ci}(t)$ = Reactor coolant inlet temperature (°F)

P(t) = Total power generated in the core (Btu/sec)

= Total thermal capacity of core materials (Btu/°F)

= Total thermal capacity of coolant in the core (Btu/°F)

= Specific heat of coolant (Btu/blm°F)

= Coolant mass flow rate (1bm/sec)

Total heat transfer coefficient (Btu/sec°F)

and

$$T_1 = \frac{C_F}{h_{FM}}$$

$$T_2 = \frac{C_M}{h_{FM}}$$

Let
$$P(t) = P_{o} + \sigma^{P}(t)$$

$$T_{F}(t) = T_{fo} + \sigma^{T}_{F}(t)$$

$$T_{AV}(f) = T_{AV} + \sigma^{T}_{AV}(t)$$

$$T_{Ho}(t) = T_{Ho_{o}} + \sigma^{T}_{Ho}(t)$$

$$T_{c_{i}}(t) = T_{c_{io}} + \sigma^{T}_{c_{i}}(t)$$

$$(B 4)$$

where the lowest zero subscript denotes a steady state value and the delta prefix a small perturbation about this value.

From substitution of equations (B 4) into equations (B 1), (B 2), and (B 3), with the reactor in a steady state at $t \le 0$,

$$T_{F_0} = T_{AV_0} + \frac{T_1}{C_F} P_0$$

$$T_{AV_0} \left(1 + 2 \frac{T_2}{T_0} \right) = T_{F_0} + 2 \frac{T_2}{T_0} T_{C_{i_0}}$$

$$T_{AV_0} = \frac{1}{2} \left(T_{H_{0_0}} + T_{C_{i_0}} \right)$$

$$(B 5)$$

and

$$d^{2}P(0) = 0$$
 (B 6)
$$d^{2}T_{P}(0) = d^{2}T_{P}(0) = d^{2}T_{Q}(0) = 0$$

Impose the perturbations at t=0. Substituting equations (B 4) and (B 5) into equations (B 1), (B 2), and (B 3) with the initial conditions given in equations (B 6),

$$dT_{F}(t) + T_{1} = dT_{F}(t) = dT_{AV}(t) + \frac{T_{2}}{C_{F}} dP(t)$$
 (B 7)

$$d^{T}Av(t) \left[1+2\frac{T_{2}}{T_{0}}\right] + T_{2} \frac{d}{dt} \left[d^{T}Av(t)\right] = d^{T}F(t) + 2\frac{T_{2}}{T_{0}} d^{T}C_{1}(t)$$
 (B 8)

$$\sigma^{T} T A V(t) = \frac{1}{2} \left[\sigma^{T} H_0(t) + \sigma^{T} C_1(t) \right]$$
 (B 9)

Taking the Laplace transform of equations (B 7), (B 8), and (B 9) with the initial conditions given by equations (B 6)

$$T_{F(3)} [ST_{1}+1] = T_{AV}(S) + \frac{T_{1}}{C_{F}}P(S)$$
 (8 10)

$$T_{AV}(S) \left[S_{12}^{2} + 1 + 2 \frac{12}{70} \right] = T_{F}(S) + 2 \frac{12}{70} T_{C}(S)$$
 (B 11)

$$T_{AV}(S) = \frac{1}{2} \left[T_{H}(S) + T_{C}(S) \right]$$
 (B 12)

where the delta prefix and lowest subscript have been deleted for readability

Substituting equation (B 12) into equations (B 10) and (B 11)

$$T_F(sT_{1+1}) = \frac{1}{2}(T_H + T_C) + \frac{T_C}{C_F}P$$
 (B 13)

$$\frac{1}{2} \left(T_{H} + T_{C} \right) \left(5 \vec{1} + 1 + 2 \vec{7} \right) = T_{F} + 2 \vec{7} \vec{7} T_{C}$$
 (B 14)

where the s domain dependence notation has been deleted for readability.

Solving equation (B 13) for T_F and substituting into equation (B 14)

After expanding the products and collecting terms

$$T_{H}\left[s^{2}\frac{\vec{1_{2}}\vec{1_{2}}}{2}+\left(\frac{\vec{1_{2}}}{2}+\frac{\vec{1_{1}}}{2}+\frac{\vec{1_{1}}\vec{1_{2}}}{T_{6}}\right)s+\frac{\vec{1_{2}}}{T_{6}}\right]=-T_{c}\left[s^{2}\frac{\vec{1_{2}}\vec{1_{2}}}{2}+\left(\frac{\vec{1_{2}}}{2}+\frac{\vec{1_{1}}}{2}-\frac{\vec{1_{2}}\vec{1_{2}}}{T_{6}}\right)s-\frac{\vec{1_{2}}}{T_{6}}\right]+\frac{\vec{1_{1}}}{G}P$$

After multiplying through by and solving for $T_H(\delta^T_{H_0}(s))$

$$\int T_{H_0}(s) = \frac{-\left\{\frac{\vec{1}_1\vec{1}_2}{2}s^2 + \left[\frac{\vec{1}_0}{\vec{7}_2}\left(\frac{\vec{1}_1}{\vec{7}_2} + 1\right) - \vec{1}_1\right]s - 1\right\}\int T_{C_1}(s) + \chi\int P(s)}{\left\{\frac{\vec{1}_2\vec{1}_2}{2}s^2 + \left[\frac{\vec{1}_0}{\vec{7}_2}\left(\frac{\vec{1}_1}{\vec{1}_2} + 1\right) + \vec{1}_1\right]s + 1\right\}}$$

where $y = \frac{1110}{912}$ and all notation has been restored.

This is equation (26) on page 28.

APPENDIX C

DEVELOPMENT OF THE HEAT EXCHANGER TRANSFER FUNCTION

The previously derived heat exchanger heat transfer equations are

$$T_{c_0}(t) = \frac{2T_3(t) - T_{H_1}(t)[1-K_1]}{[1+K_1]}$$
 (C 1)

$$\frac{C_S}{h_{TM}} \frac{dT_S(t)}{dt} + T_S(t) = \frac{1}{2} \left[T_{H_1}(t) + \overline{I}_{C_0}(t) \right] + \frac{1}{h_{TM}} P_L(t) \qquad (C 2)$$

where T_{Hi}(t) = Heat exchanger coolant inlet (°F)

 $T_{Co}(t)$ = Heat exchanger coolant outlet (°F)

 $T_{\varsigma}(t)$ = Saturated steam pressure (PSIQ)

 $P_{I}(t)$ = Power delivered by heat exchanger (Btu/sec)

Cs = Total thermal capacity of heat exchanger
 metal and secondary water and steam (Btu/°F)

C = Specific heat of coolant (Btu/lbm°F)

h_{TM} = Total heat transfer coefficient (Btu/°F.sec)

m_m = Coolant mass flow rate (1bm/sec)

and $K_1 = \frac{2m_MC}{h_{TM}}$

Let
$$T_{Hi}(t) = T_{Hi_0} + \sigma^t T_{Hi}(t)$$

 $T_{Co}(t) = T_C + \sigma^t T_{Ho}(t)$ (C 3)
 $T_S(t) = T_S + \sigma^t T_S(t)$
 $P_L(t) = P_{L_0} + \sigma^t P_L(t)$

where the lowest zero subscript denotes a steady state value and the delta prefix a small perturbation about this value.

With the plant in an initial steady state at $t \le 0$, equations (C 1) and (C 2) are

$$T_{C_{O_O}} = \frac{2T_{S_O} - T_{H_{1_O}}[1 + K_1]}{[1 + K_1]}$$
 (C 4)

$$Ts_o = \frac{1}{2} \left(T_{Hi_o} + Tc_{oo} \right) + \frac{1}{h_{TM}} P_{Lo}$$
 (C 5)

and

Impose the perturbations substituting equations (C 3), (C 4), and (C 5) into equations (C 1) and (C 2)

$$\int T_{C_0} = \frac{2 \int T_S - \int T_{H_1} (1 - \kappa_1)}{(1 + \kappa_1)} \tag{C 7}$$

$$\frac{C_S}{h_{TM}} \frac{d}{dt} \int_{S}^{T_S(t)} + \int_{S}^{T_S(t)} = \frac{1}{2} \left[\int_{H_1(t)}^{T_{H_1(t)}} \int_{C_0(t)}^{T_{C_0(t)}} - \frac{1}{h_{TM}} \int_{C_0(t)}^{P_L(t)} \right]$$
(C 8)

Taking the Laplace transforms of equations (C 7) and (C 8) with the initial conditions given by equation (C 6)

$$\frac{dT_{5}(s)}{dT_{5}(s)} = \frac{\frac{1}{2} \left[dT_{C_{0}}(s) + dT_{H_{1}}(s) \right] - \frac{1}{h_{TM}} dP_{L}(s)}{T_{5} s + 1}$$
where
$$T_{5} = \frac{C_{5}}{h_{TM}}$$

$$dT_{C_{0}}(s) = \left\{ 2 dT_{5}(s) - dT_{H_{1}}(s) \left[1 - \kappa 1 \right] \right\} / \left[1 + \kappa_{1} \right]$$

These are equations (34) and (35) on page 33.

APPENDIX D

DEVELOPMENT OF THE PRIMARY PIPING TRANSFER FUNCTIONS

1. Transport Delay Transfer Functions

The transport delay of the coolant between the outlet of the reactor and the heat exchanger inlet plenum can be expressed as

$$T_{\text{Hi}_p}(t) = T_{\text{Ho}}(t-T_3)$$
 (D 1)

where

T_{Hi} (t) = Heat exchanger inlet plenum coolant
 temperature (°F)

 $T_{Ho}(t)$ = Reactor coolant outlet temperature (°F)

73 = Transport time delay (sec)

Rearranging terms and expanding equation (D 1) in a Taylor series

$$T_{H_0}(t) = T_{H_{ip}}(t) + T_3 \frac{dT_{H_{ip}(t)}}{dt} + \frac{T_3}{2!} \frac{d^2T_{H_{ip}(t)}}{dt} + \dots$$
 (D 2)

For slow temperature changes, the second order and higher terms in equation (D 2) may be ignored.

$$T_{H_{\frac{1}{2}p}}(t) + 13 \frac{dT_{H_{\frac{1}{2}p}}(t)}{dt} = T_{H_0}(t)$$
 (D 3)

Let

$$T_{Hi}(t) = T_{Hi_{po}} + \int T_{H_{ip}}(t)$$
 (D 4)
 $T_{Ho}(t) = T_{H_{oo}} + \int T_{Ho}(t)$

where the lowest zero subscript denotes a steady state value and the delta prefix a small perturbation about this value. From substitution of equations (D 4) into equation (D 3) and for the temperatures in a steady state at $t \le 0$

$$T_{H_{\hat{1}_{DO}}} = T_{H_{O_{O}}} \tag{D 5}$$

and

Impose the perturbations at t=0. Substituting equations (D 4) and (D 5) into equation (D 3) with the initial conditions given in equation (D 6)

$$\int T_{H_{ip}}(t) + T_3 \frac{d}{dt} \int T_{H_{ip}}(t) = \int T_{H_0}(t) \tag{D 7}$$

Taking the Laplace transform of equation (D 7) and solving for $d^{\rm T}_{\rm Hi}$ / $d^{\rm T}_{\rm Ho}$

$$\frac{\partial^{7}H_{1p}(s)}{\partial^{7}H_{0}(s)} = \frac{1}{1+\tilde{t}_{3}s}$$

This is equation (39) on page 36.

By following a similar development, the transport delay between a perturbation in the heat exchanger coolant outlet temperature and the resulting perturbation in the reactor inlet plenum coolant temperature can be expressed by the transfer function

$$\frac{\partial^{1}Tc_{ip}(s)}{\partial^{1}Tc_{o}(s)} = \frac{1}{1+\overline{1}45}$$

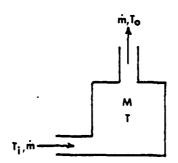
where of Tc_{iP} = Reactor inlet plenum coolant temperature perturbation (°F)

 $\mathbf{1}_{A}$ = Transport time delay (sec)

This is equation (40) on page

2. Mixing Delay Transfer Function

Let the reactor and heat exchanger coolant inlet plenums be represented by the control volume shown in Figure D-1.



M = Mass of coolant in plenum (1bm/sec)

T = Average plenum coolant temperature
 (°F)

To = Plenum outlet coolant temperature
 (°F)

m = Coolant mass flow rate (1bm/sec)

Figure D-1. Mixing Volume

If perfect mixing is assumed, the coolant outlet temperature is equal to the average coolant temperature in the plenum. At heat balance on the volume requires that the net heat flowing into the plenum be equal to the increase in stored energy. Assuming no ambient heat losses, and assuming the temperature of the plenum structural material remains constant, the heat balance per unit time over a time interval Δ t is

$$\dot{m} \Delta t C \left[T_1(t) - T_0(t) \right] = MC \frac{dT(t)}{dt} \Delta t = \dot{m} \Delta t C \frac{dT_0(t)}{dt}$$

where C = specific heat of coolant

The resulting equation is

$$\frac{M}{\dot{m}} \frac{dT_0(t)}{dt} + T_0(t) = \vec{1} \frac{dT_0(t)}{dt} + T_0(t) = T_i(t)$$
 (D 8)

where $T = \frac{M}{\dot{m}} = mixing time delay.$

Let

$$T_{i}(t) = T_{io} + \int T_{i}(t)$$

$$T_{o}(t) = T_{o} + \int T_{o}(t)$$
(D 9)

where the lowest zero subscript denotes an initial steady state value and the delta prefix a small perturbation about this value.

Substituting equations (D 8) into equation (D 9)

$$\mathcal{T} \frac{d}{dt} \int T_0(t) + \int T_0(t) = \int T_1(t) \tag{D10}$$

Taking the Laplace transform of equation (D10) and solving for ${\it d}\tau_{o}/{\it d}\tau_{i}$

$$\frac{\mathbf{d}^{T_0}(s)}{\mathbf{d}^{T_i}(s)} = \frac{1}{1+T_s}$$

This is equation (41) on page 37.

APPENDIX

LISTING PROGRAM INTERACTIVE

The state of the s CM2) 9,78,79, ,51,53,32,65,84,79,77,83,47,67,77 -5/, MODIM/6280./, LOADO, ũ BTU/LAM 200 SECTION SECTION X E ,LAB1 (29), LAB2(21) TY (,32 S S S S E FOLLOWING PLANT PARAMETERS:

UAAVIUM-235 412R0SCOPIC A3SORPTION CROSS

UAAVIUM-235 412R0SCOPIC A3SORPTION CROSS

ESTATEMENT PARAMETERS:

UAAVIUM-235 412R0SCOPIC A3SORPTION CROSS

UAAVIUM-235 412R0SCOPIC A3SORPTION CROSS

UAAVIUM-135 DECAY

UAAVIUM-135 DECAY

UAAVIUM-135 DECAY

UAAVIUM-135 DECAY

UAAVIUM-135 CAAVIUM AASS FLOM AATE (LBM/SEC)

REACTOR COOLANT SPECFIFC THERMAL CAPACIT

STEAM FEVERATOR TOTAL EFFECTIVE HEAT TRA ں۔و ~0355 8 5,84,68,79,87,78 WU F PER .LAMDAX/2.3917E X(3000), TI YE(3000), TI MEH(3000) 2,86 2,86 POWER -51 INTRODUCTION STATE LE: 3 D S T R A N S 11 82,65,78,83,73 LE: K E N D /2.8 FEADY PWRO 13.KI.LAYJAI - 6.0776+15 - 393.67-24 - 334.77-24 - 1.56256-1 - 0.005286 1.16 5919. AL 51 RDGRAY 1.50000 SCALE TABLE TO SECALE TABLE TO SECALE TABLE TO SECALE TABLE TABLE SCALE TABLE TOURLESS OF TOUR S TOUR STREET TOUR STREET S 14111 16+6 5+500 COAMANA HALL RINT PR EAD INEAD * 78 3 2 SCALE SCALE STALE STALE STALE Z $\alpha \mathbf{Z} \alpha$

71

0 0 0 00000000000000

UU UUA

72

ပပ

Ť

ပပ ပ

(MIHSO) REMEMBER (TSHIM) MOVEMENT 6 S ENTRY RIJO INVENT EVTRY EVTRY FRAMINE IF INCORRECT DATA ENTERED (TSHIM *GE* 0 * AND. TSHIM *LE. 10. PAINT NOTICE OF INCORRECT DATA ENTRY WAITE (6.6200) CONTROL CONTROL DATA ⋖ 13 35 130 13 65 10002855 DATA WINE INTITAL REACTIVITY CHANGE (6,5004) DAT ENTERED TRANSFER TSHIM TO INITIAL DATA FILE WRITE (1,1000) TSHIM ENTERED ENTERED 17 55 1 13 50 IVCOXXECT TO 30 60 TO 80 INCORRECT 84NK 3 AVK SI MUL 4 TI 3 V **JAT4** DATA DATA F. 1ME LENST4 (SEC) DF (6,6005) TSHIM DETERMINE IF INCORRECT DA ISHIM * ED. 1) GO TO IF (SHIM * ED. 1) GO TO IF (SHIM * ED. 1) GO TO IF (SHIM * ED. 1) GO TO IF (SHIM * ED. 1) GO TO IT (SHIM * ED. 1) GO DETERMINE IF INCORRECT D IANS = IFIX(A4SAER) IF (IANS * EQ* 1) G3 T3 IF (ANSWER * EQ* 2) SRINT NJTICE 3F IN MRITE (5,5200) DETEXMINE DESIRED SIMULA
READ (5.5000) ANSWER
DETERMINE IF NCORRECT
TANS = (FIX (ANSWER)
IF (IANS = (FIX (ANSWER)
IF (IANS = 2) 62
SRINT NOTICE OF IN DETERMINE THE DIRECTION WRITE (6,6006) READ (5,5000) SHIM READ TEREAD RITE FAD

1

CONTRO BANK MUUUTH) AEACTIVITY ("IN") OF POST STUTDOWN XENON TRANSIENT ALGORITHY
CONTINJE
CALCULATE EQUILIBRIUM IDDINE-135 AT TIME OF SHUTDOWN
IO = GAMMAI * 1235 * SISMAF * FLUXO / LAMDAI
CONVERT LAMDAI AND LAMDAX TO 1/MIN(
LAMDAI = LAMDAI * 60. ш ENGTH (TS-114) AMD DIRECTION (DSHIM) FIL 63 REACTIVITY DETERMINE IF INCORRECT DATA ENTERED

IF (LOADF .GE 10. AND. LOADF .LE. 100.)

PRINT NOTICE OF INCORRECT DATA ENTRY
WAITE (6.6230)
GO TO 70 SET FINAL TURBINE LOAD = INITIAL TURBINE LOAD | DATA TRANSFER TSHIM AND DSHIM TO THE INITIAL WRITE (1,1000) TSHIM WRITE (1,1000) DSHIM SET SHIM DIRECTION TO INSERT POSITIVE DSHIM = 1.60 TO 50 SET SHIM DIRECTION TO INSERT NEGATIVE D'SHIM = -1. TRANSFER DSHIM TO INITIAL DATA FILE WRITE (1,1000) DSHIM FILE FILE TRANSFER LOAD= TO INITIAL DATA WRITE [1,1000] LOADF GO TO 200 READ FINAL TURBINE LOAD (LOADF WRITE (6,6007) LOADF READ (5,5000) LOADF TRANSFER LOADF TO INITIAL DATA WRITE (1,1000) LOADF GO TO 200 SET THE TIME ROD MOVEMENT : TSHIM = 0.

2 2 2 3 3 4 3

೧೯ ೧೦ ೧೦

200

ပပစ္ထပ

```
CALCULATE INSTANTANEOUS XENON-135 CONCENTRATION

TIME (T) = F10AT(T-1)

X2 = EXP(-L40DAX*TIME(T1))

X3 = EXP(-L40DAX*TIME(T1))

X3 = EXP(-L40DAX*TIME(T1))

X3 = EXP(-L40DAX*TIME(T1))

X4 = T10 (L40DAX*TIME(T1))

X5 = EXP(-L40DAX*TIME(T1))

X6 = L40DA1*TIME(T1) - EXP(-L40DA1*TIME(T1))

TIME(T1) = T1 = T2 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + 10.00

TIME(T1) = T1 = T1 = T0 + T0 = T0 + T0 = T0 = T0

TIME(T1) = T1 = T1 = T0 + T0 = T0 = T0 = T0

TIME(T1) = T1 = T1 = T0 + T0 = T0 = T0 = T0

TIME(T1) = T1 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0

TIME(T1) = T0 = T0 = T0 = T0

TIME(T1) = T0 = 
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             A. GORITHM
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                32)
00, L3, L483)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  GRAPHICS
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        10111(12)
BINIT(12)
BINIT(12)
NPTS(300)
STIMK(100,970)
STIME (100,100,100)
STIME (100,100)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  401
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          100
                                                                                                                                                                                                                                                            82
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         ၁၆
                     ပ
```

11111111111111111111111111111111111111	100840000000000000000000000000000000000	INA 0273	NA0249 NA0280 NA0280	1 N A O C B C A N A O C B C A O C B	20000000000000000000000000000000000000	INA0291 INA0292 INA0293	NA 0294 NA 0296
CNA	AIZHZ HUQU	マリョ	RAT			- Gr.	
. X	H ALSSA		OFF	SS	C	. •0	
Z α	HZWODKKHO		\cap I		1-0-0		
1 O	X MZ3- NMQGN⊬M F	HIN.	EN		ころはしている。	~ O	
⊢ ∝	I>40&mv ↔	7 ~ ₩ ⊢	-	I +	-Z ¥ ⊦		
α <u> </u>	0110 02/0	2/>&C	1. 0.5		AHOU CONTRACT		
ION		アースの	- 0		`F0 >F∞α	•~	
Ψ ⊷	IHO · NZU	> 444	×	ш >	とートしょう	• oo	
A A	4 +- NOUMOO		2.5	シック	STASS-		
C ~	ペース トーニコ	LOUG	\•	∀ ⊢0	o - -	- 7-	
ָט <u>ה</u>	0 Z = 1 C C C C C C C C C C C C C C C C C C		- ×	UNU	ANSON XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX		
S E	₹₹₽╱╟ ⊙• Z	α	O	⊃ 0	Suint C	> 3 4	
Ĩ	∞α- шш/~ (ZZ	0	-	• × ×	· LIF	
11 H	>-32X /K]	E THE	H•		10 HA-		
∨ ⊢		E Z	< ∞	-20	ი∝ — •თ ა	200	
ш « Т ш ».	トトロレイア・心	اجري	الة 14				
⊢ ⊢ ⊢ Z	⋖ ≠• ∨∩ ⊸∋.	11114 H-	ァつ	4X]	[C - C	- LL)	•
AX AX ADF TID	ほし/女に *C o	CHH N	S	A *C	-&C + 20	111	ED
A A A A D	FW +0120U1	ZV4.	AT				A.
	H +Z>4 HU	→…∟	. .	₹•_	ことメノアア	• •	-
W I OF FU	Z & -UICH 4-1-KU	ZUUZ	₩()	—————————————————————————————————————	1大 E) ト・C	. 33	N N
3 • m= mm	4470>4-22		SA	レフス	ころくとなと	- ⊢ Z	
O X NY ZX	0 -0-40 00-40 10-40		90 98		- A - A - A - A - A - A - A - A - A - A		TA
87X I • WIL	70IIIn		шш	U1 U	→		40
നനം ഗമയ_	いくロエルペープ	ニコアへな	TIL	IOI	こころとは、これに		
	H # OACO		Z	- C-	-F20-4	A C	<u>د</u> -
AXE HO SH	と 日本 ア 日本	- e •u	AC.	oc 0	ヘムオチャ	(Z~>	w
41 · > · I>	Z∝ ⊢m4	つトブロ	HΨ		4 W-U	 □	8
NIC FO S	0 0 00+0	SEE SEE	ZX		こくしょきていい		00
3×3 U3 FT	いんて ・まかい	C AE	Ω.	~ Z	+mr0 +	-w _,	ž
04	・ペイータフロチト	-C- UI	•	Y - 1	- OM- 4	7	-
7 0 4 7 E	マー・マー・マー・マー・マー・マー・マー・マー・マー・マー・マー・マー・マー・マ				、ゲ ナル ヘル 、		•
ICO 10 0	NON ADZ NHA	\O +	5×4:	>\ <i>Z</i> \	XW \ -	· > \0	
1>0 H0 MIN 0 4 4 Amo	880 47 ZOUN + •••≥ШHШШ ⊢				F 00 = >		•
#F +0 F + NF + E		് വ	% 41	LI.		- د	_
トレーラ まし うら り	しししトコーベエ ナロ	ころしがつ		— .~	~-		_
	FFF NYDUZHO	04 D	-4T	⊃∢- <	【!!!!~!!! !	4 C C -	-<
	IZZIUH ZWO"	1 7 ·-	×≠u∶	Z5 +1	「ナーファ。	m2F	• Σ
NAME OF THE PARTY	RARLACACHTA	*OO	∢ ○ :	XOSC	>>		wc
THE SEAS OF THE	******* •••••••••••••••••••••••••••••	~ ₩ ₩	$\mathbf{L}\alpha$	ᄔᄥ	この女とよら	JEHU	SIL
			_			_	_
0 0	900 900	002	03	0 4	9	0 7	00
	000 000	90	909	600		9	62

```
CONTROL OF STATE OF S
                                                                                                                                                                                                                                                                                                                                                                                                          36i
                                                                                                                                                                                                                                                                                                                                                                                                              N
N
N
          CUTT
WÉTERS: 1/5
2 SEC: 1/5
FINAL TURBI
                                                                                                                                                                                                                                    A¥
TO⊤
                                                                                                                                                                                                                                                                                                                                                                                                              MUL A
                                                                                                                                                                                                                                      u «
                                                                                                                                                                                                                                                                                                                                                                                                              6500
        6301
                                                                                                                                                                                                                                    6400
                                                                                                                                                                                                                                                                                                             640]
```

man was be the work

APPENDIX

PROGRAM

S4C1639. TIME=90 , " HEATH, 6.6. //REACTOR1 JJB (1085,1077,4783) // EXEC CSMPXV //FT13FJ01 DD SPACE=(5YL,(5,2)) //X.SYSPRINT DD DJMMY //X.COMPRINT DD DJMMY //X.SYSIN DD *

THIS PROGRAM SIMULATES A HIGHLY ENRICHED RESSURIZED WATER REACTOR.

THE APPLICABLE P. ANT PARAMETERS FROM THE SHIPPINGPORT ATOMIC POWER

STATION ARE USED IN THE MODEL. THE THERMAL RESPONSE OF THE PLANT

TO NORMAL OPERATING TRANSIENTS, INDUCED BY EITHER CONTROL ROD OR

TURBINE LOAD CHANGES, ARE SIMULATED. THE PROGRAM ALSO INCORPORATES
A ROD CONTROL, REACTOR PROTECTION, AND AUTOMATIC AVERAGE REACTOR

SDFLAG, CFLA31, CFLAGZ, CFLAG3, CFLAG4, THFLAG, TFLAG, SFLAG1, SFLAG2 T, T1, T2 FIXED

TIAL

TEMPERATUR COOL ANT REACTOR AVERAGE HH ZERO COUNTERS (T1, T2) FOR (TMRT) CONTROL ALGORITHM:

ZERO THE FOLLOWING FLAGS (INDICATORS) FOR THE REACTOR PROTECTION TWRT CONTROL ALSORITHMS (THE FUNCTION OF THESE FLAGS WHEN SET IS SCRIBED, WHEN ZEODED THEY ARE NEUTRAL):

SCRIBED, WHEN ZEODED THEY ARE NEUTRAL):

CFLAGS PROTECTIVE FULL INSERTION OF CONTROL ADDS (SCRAM)

CFLAGS SCAM ON REDCCJAING CUTBACK

CFLAGS ZEO CFLAGI

THFLAG HISH REACTOR COLLAGI

THFLAG HISH REACTOR COLLAGI

THLAG HISH REACTOR COLLAGI

SFLAGS ZEO SFLAGI

SFLAGS ZEO SFLAGI

THAN TO UTSTOE NORMAL TEMPERATURE ALARM

SFLAGS ZEO SFLAGI

AND

CFLAG1=0,CFLAG2=0,CFLAG3=0,CFLAG4=0,THFLAG=0,FLAG1=0,5FLAG2=0,THFLAG=0, G=0+7 =0+5F SD=LA(TF:A3 INCON

```
STORASE FOR DYNAMIC SECTION TAANSFER FUNCTION COEFFICEINTS:
3),0(3),E(2),F(3),3(2),H(2),I(3)
                                                                                                                                                                                                                                                                   FOLLOWING PLANT CONSTANTS:
URANIUM—235 NUMBER PENSITY (ATOMS/CM3)

EFFECTIVE DELAYED VELTAN FACTION (DELTA K)

EFFECTIVE FISSION YIELD OF FOODNEL 35

EFFECTIVE FISSION (CRSTANT (1/SEC)

AN URANIUM—235 MICROSCOPIC ABSORPTION CROSS SECTION (CR2)

AN URANIUM—235
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   5, 5 16 A A X = 1.5626 E - 18, 1 = 1.16, 1 A U 4 = 3., 1 A U 4 = 3., . . .
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         -3.1E-4, 5, 5, DJ4MY=0.0
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  U235=5.07E+16, BETA=77.0E-4, COMPANIE C
                                                                                                                                                                                                  EST THE COURT OF T
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  PHHAMIN
CCCCOAR
DCCCCAR
DCCCCAR
DCCCCAR
DCCCCAR
DCCCCAR
DCCCAR
DCCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCAR
DCCCCAR
DCCCAR
                  ESTABLISH
STORAGE C(
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         PARAMETE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              CONSTANT
```

```
FOLLOWING INITIAL STEADY STATE PARAMETERS:

INITIAL REACTOR POWER (BTJ/SEC)

INITIAL THERMAL NEUTRON FLUX (NEUTRONS/CM2 SEC)

INITIAL EDUILIBALUM XENON-135 NUMBER DENSITY(ATOMS/CM3 INITIAL BOUNDER STEAT IN TEMPERATURE (F)

INITIAL REACTOR COOLANT JULET TEMPERATURE (F)

INITIAL STEAM GENERATOR COOLANT INLET TEMPERATURE (F)

INITIAL STEAM GENERATOR COOLANT INLET TEMPERATURE (F)

INITIAL STEAM GENERATOR COOLANT SUTTENTED TEMPERATURE (F)

INITIAL STEAM GENERATOR COOLANT SUTE TEMPERATURE (F)

INITIAL SATURATED STEAM PRESSURE (LBF/IN2)

INITIAL SATURATED STEAM PRESSURE (LBF/IN2)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         +3
7
+J235+SIGMAF*FLJXD/(LAMDAX+SIGMAX#FLUXD)
IGMAA/SIGMAX+U235)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               ROD MOVEMENT (TSHIM)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               (DSHIM
                                                                                                                                                                                                                                                                                                                                                                                                                                                                              ROCEDURE PWR3, TSHIY, DSHIM, LOADF = READIN(334MY)
READ INITIAL STEADY STATE POWER LEVEL (PWR0) (%)
READ (5,5000) PWR3
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    ROD MOVEMENT
          PLANT PARAMETERS
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               THE TIME LENGTH OF BANK CONTROL (5,5000) TS-11M
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  READ THE FINAL TJABING LDAD (LOADE)
READ (5,5000) LOADF
NOPROCEDURE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    BANK CONTROL
A EMAINING CONSTANT F

A = 1./(WDOTM*CPM)

= CM/(WDOTM*CPM)

= CF/4F4

= CF/4F4

= CM/4=M

= CS/4TM

2. * MOOTM * CPM/HIN
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             PARO
                                                                                                                                                                                                                                                                                                                                                                  CPM/HT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              MX S
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   PWRJT = PWRJ * 2.19E
FLJXO = 2.5+12 * PWR
XJ = (GAMMAX+GAMMAI)
ALPHAX = -1./(XJ + S
TMRJ = 523.
THJJ = TMRJ + 15 *
TCJJ = TMRJ - 15 *
TCJJ = THJJ
TCOJ = TCIJ
LJADO = PWRJ
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         THE DIRECTION OF (5,5000) DS4IM
     TTACUST TACUST T
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    APACTATATAX PER DONOLUS A TOUNDOUND A TOUNDOUND OUT A TOUNDOUND A TOUND A TOUN
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          CALCULA
          CAL CUL
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               READ
READ
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    READ
READ
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             ٩
```

```
CALCULATE DYNAMIC SECTION TRANSFER FUNCTION (TRANSF) COEFFICIENTS:

C(1) = TAJO/2 * (TAJI/TAU2 + I.) - TAUI

C(2) = TAJO * TAUI/2.

C(3) = TAJO * TAUI/2.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(9) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(9) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = -1.

C(8) = -1.

C(9) = -1.

C(1) = -1.

C(1) = -1.

C(1) = -1.

C(1) = -1.

C(2) = -1.

C(3) = -1.

C(4) = -1.

C(5) = -1.

C(6) = -1.

C(7) = -1.

C(8) = 
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              PRINT INITIAL P.ANT PARAMETERS
WRITE(6,5000) PWRD,TMRO,THOO,TCIO,TSO,PS),LOADO,DSHIM,TSHIM,LOADF
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          CALCULATE THE TJ23INE LDAD CHANSE TIME (TLOAD)
RLDAD = 3.005
TLOAD = ABS(LOA):1*2.0
                                                                                                                                                                                                                                                                                                    ₩
•
TSO = ( 7HI) * (1.-KI) + TSOO * (1.+KI)
PSJ = EXP((4.3G(TSJ)-S2)/SI)
AD = LOADO/100.
                                                                                                                                                                                                                                                                                      (LOADC)
                                                                                                                                                                                                                                                                                                                                                                                                                                      SET THE DIRECTION OF TURBINE LOAD CHANGE IF ( LOAD 2 1 5,10,15 TURBINE LOAD CHANGE GO TO 17 0000 RLOAD = 0.0 RLOAD = 0.0 GO TO 20 0000 RLOAD = -1.
                                                                                                                                                                                                                                                                                               CALCULATE THE TJABINE LOAD CHANGE LOADC = LOADS - JADF
                                                                                                                                                                                               TACSON
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   10
```

```
410N
COEFFICEINT
            TIMES
             CONTROL
FUNCTION
             300 AND TJABINE
    BET
TRANSFER
INETICS TRANSFER
            1 = 0.5
2 = TS+1 4
5 = 0.5
¥#####
                    ٥
                    ローと123
             MAEL
DAC
OTTHHH
4
                                                          CHOOSE
             ABLI
TIV
u.
                                                           ----
                    œ
                    MAJOI
             EST.
          SORT
```

```
(BTU/SEC)
                   NE (%)
URBINE
ETTINS
TO TURBIN
TO THE TU
VALVE SE
POWER DELIVERED TO THE TARGET TO THE THROUGH THE THROTTLE
                                                                                                                        INSTANTANEOUS
INSTANTANEOUS
INSTANTANEOUS
PWRSG
PWRSGT
A
                                                                                                                        ROD
               DYNAMIC
*
                                                                                                                                    TUR
```

```
## REACTOR POWER STARTUP RATE (SJR)

SURED FRECETIVE POWER STARTUP RATE (SJR)

SURED FRECETIVE POWER STARTUP RATE (SJR)

** REACTOR PROJECTIVE POWER STARTUP RATE (SJR)

** CHECK IF SURED FRECETIVE PROJECTIVE P
, CFLAG2,
OE3
                  0110
               PROCEDURE
```

```
STOP CUTBACK
TIME! = TIME
TIME2 = TIME
SET CELAG2 (REDCCURING CUTBACK WILL RESULT IN SCRAM)
SET CELAG3 (ZERO CFLAG1)
CFLAG3 = 1
CFLAG3 = 1
CFLAG3 = 1
                                                                                                                                                                                                                                                                        CLEARED
                                                                                                                                                                                                                                                                                                                 CHECK IF HISH SUR CUTBACK CONDITION CLEARED
IF (SUR GE 1:2) GO 13 115
PRINT NOTICE HIGH SUR CUTBACK CONDITION CLEARED
WRITE(6,6008) POWER, SUR, TIME
                                                                                                                                                                                                                                  CHECK IF HIGH PDAER CUTBACK CONDITION CLEARED IF (CFLAS "EQ. 1) GO TO 50 IS IF (PDWER "GE 100.) GO TO 115 PRINT NOTICE HIGH POWER CJTBACK CONDITION WAITE (6.5007) POWER, TIME GO TO 55
                               PRINT NOTICE HIGH SUR CUTBACK CONDITION WRITE (6.6009) POMER, SUR, TIME CELAG4 = 1 GO TO 70
 SCRAM
NOTICE HIS4 TEMPERATURE (6,6006) POWER, TH, TIME
    PA I NT
WA I TE
                                    SET
              M***4*
                                                                                                                                                                                                                                                                                                                                                                                                                                                                       ę
**
```

```
CHECK FOR TMR CONTROL CONDITION

IF (TFLAS .EQ. 1) T1=T1+1

CHECK IF TMR IS 09TSIDE NORMAL OPERATING TEMPERATURE (NOT) BAND

IF (ABS(TMR) .LE. 5. ) 53 T3 110

IF (TFLAS .EQ. 1) G3 T3 95
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         AVERAGE REACTOR CODLANT TEMPERATURE (TAV) CONTROL SYSTEM ALGORITHM
                                                                                                                                                                                                                                                                                                                                                                                                                                     PRINT NOTICE REDCCURING CUTBACK RESULTING IN SCRAM WRITE (6, 6011) POWER, SUR, TIME 63 TO 40
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 BAND
                                                     CHECK FJR REDCCJYING CUTBACK

SET CJTBACK PARAMETERS:
SET RDD DIRECTION "IN"
DSHIM = _1.
SET TIME TD START CUTBACK
TIMEI = TIME + 370.
SET CLFASI
SET CLFASI
FRIP TURBIVE THROTTLE TO MID POSITION
IF ( AD °LE. 0.5) GO TO 75
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      CHECK IF THOT ALARM CONDITION CLEARED
IF (TH .3E. 544.) GO TO 90
REZERO THFLAG
THFLAS I L
PRINT NOTICE THOT ALARM CONDITION CLEARED
WRITE (6,6002) TH, TIME
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  (TFLAS EQ. 1) G3 T3 95
SET TFLAG (TMR 03TSIDE VOT BAND)
TFLAG = 1
SET TIME WHEN TMR WENT OUTSIDE VOT
TIMET = TIME
PRINT NOTICE HISH POWER CUTSACK CONDITION WRITE (6,6010) POWER, TIME
                                                                                                                                                                                                                                                                                                                                         DLOAD = -1.
SLOAD = TIME
TLOAD = TIME
GO TO 115
                                                                                                                                                                                                                                                                                                                                             75
```

```
## CALCULTE THE AMJUNI OF TIME TWR DUTSIDE NOT BAND
## CALCULTE THE AMJUNI OF TIME TWR DUTSIDE NOT BAND
## CALCULTE THE AMJUNI OF TIME TWR DUTSIDE NOT BAND
## CALCULTE THE AMJUNI OF TIME TWR DUTSIDE NOT BAND > 10 MINUTES
## CALCULTE THE AMJUNI OF TIME TWO TO THE TO THE TWO THE CALCULTE THE TWO THE CALCULTE THE TWO THE CALCULTE THE TWO THE T
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       5 CONTINUE
ENDPROCEDURE
                                                                                                     100
```

	" Vo" -" -	<u>د</u> د		۰ ۵	α: -	4 F	S			_		
	**************************************			-	1	>	•	Δď	HE PE	E E	-X	
	1,01,51,8		Z =	r O S	m2	mm ⊢	1	E.O	F-S	- -	•••	
	α •αΨ+Z •			٩Z	>0		•	-	Ö •	< ▶	\	
	3FF3 *** *	ARM ARE	Z>	Z ()	ALEC	DIT	ΕD	₩ ¥.	Ξ.V.	>×	- 86	
	G OFXF		. m.⊢	SS	>0	20	AS	BA 9	+:	ĽE '	E0	
	emac emo	A L			U.	Ξœ	UI-	W	UI 🗢	-	F~~	
	Q •• ₫ •∺Q		S N X	∵•	TL 2,	7		OLU -	Z·	⊢•	₽ 9 S	
	-6 •Σ• 00	Ex	. JUT	~	-	$\mathbf{u}_{\mathcal{O}}$	×c	2 11	œ=`	Ŧ.	F#0	
	A *UF IN S	I	E 🖚	ED 6.	R0 F6	HEN-	AC	10C	22	12.	Z.O	
	& # / ₹₫±=		SE-	α·	I.	∞~•	Θ (と と と に な に	⊢ •€	\$ m\$	U ≡ U	
	- W - W -	_	• >w	⊃•	10	□ → 0	ء⊂د	nd	- LL:	ZZL	対臣の	
	XDEF * * *	₩ ≠ (II)	TAP	₩ # #		×	70	<u>₽</u> Ë.		J₩ •	AD.	
	しょう くろくん		853	αX	-		٠.	~ن ⊢	- # L	TICK II	2 P C	
	\X •00 • •	404	C Z	ш Г	α. - -		VIII.	A Z	OH,	SE SE	8. 6.	
	● □ ◆ ▼ ◆ ◆ Ⅱ	<u>~</u> ∪~	こりり	A.•	EX •		α	STOP	MZ.	. 3 	• OIL	O.R.
	··▼ø∢• • ⊞	arva		Ŀ×	3	Q -	أبلله		AQ.	7 + C	,t.	CTC
	マモ・フロ・クロ・クローク 日子 スコース・コート オー	エゴ	. 4 . 47	~~ ~				XOF		~~	• ∢ ⊪	ر 2
	H - HKHH	F	- NIII	- 11 -	•	← → √	n u	とりら	-0	T - C	アピラ	w
	一直と言うといる。		T TO	^=-	66			FC -		Sin.	H +	œ
	HING CAN	m\0 m	- & - U	<u> </u>	- C	7-1		×-		E A II	HX=	E E
	~ ~ m⊢×~ ∨	- · -	- 2	ວ່ - ຫ	240	II U	LI	س€	نهن:	Σ \sim α	$\triangle \bullet \times$	
	ADANGH-		リルトロリング	シアペ	4 - 9			ΛΦ ~	4 • D		Z \ 0	ZII
	ピーシェルス	→ ₩	Σ⊲	- ~ • ₩		431	-ñ	57	Ex	- 7×	∵ • ∨	
	HAG -416	みとして	128	-CH] Z = 1	പ	•	~ ∪•0	⊃() +>	× +	XM.	Q • •
	マンド らな・	- 1-	1 -4 1	44 1- C	~ ~	\(\mathcal{O}\)	<~.	• π. επ. • C	> \ \	~~	VAA	17 14 14
	N4 - X-	~ 0 ~ 0)× 11	> # 0	KZW	ШΟ	-4-	(> •	V	B-C	or
	>×WUNU =	*CXC) ~ (3)	*O>2	-4 Z	%L`	• -	-4 II U	U47 040.	~~~~	⊃ ⊢•	SAOA
	ABT-04 +0 0 1 • F • F > ABC = :	u •	XF 1		الناقة ا	₽¥.	Ξ	14K	i i		C +	\sim 00
		-0-0	- X+	# C) F \	<i>⊃</i> 0,∝	∞ ∝	U	120 120 120 120 120 120 120 120 120 120			• <3∙	ω • • • • •
	NA - F - I		. T. III	ΣW • C	350	ш	· 64	- -	3 <u>~</u> 2.	. S.S.	PAN	<u> </u>
	ш⊢#∢₩⊣	AA A	r = 01	- w	K 3E CO	EXC	13 I	1 day	040	·IA	24> •	AZZ
	4444	∩Ш• ц	COLINA	2001	10.111					, , ,	70 -	P 3
	No XHON TO NE	~_	.47	×4	<u>-</u> -~		. —	ي ع		^_=:	الاستان	0
	まれば ない ない は ない は に な に に に に に に に に に に に に に	No oc	E E	-110	40.	O	, (C) C	romi	iiO	~ €	70 - 4	ZOC
	・25008・11		:-> _. .	\>73 ~440	5±×	#8 .	UT.	/ <u>#</u> ~3		: - C.	AK HOW	PXX DII
	~ ~~ ~ ~	~	11 7	- M	5-5	0.0	צ כ	⊃ • →	- L	A)	• TQ	
	L- +CX-) "·	,_~ 	- (n	7 - C		J• ⊦ - -	×-		D'	7-FD	TCC AII:III
	F-FWU- A	に口に	EHE	・レフ・	-	⊢ ~ <	4H-4	4 →	<	×-a,	∪™™	ZXX III
	\$2 -MF -7	4 I H I	:- УШ	ひみてく	人とこ	₹ 8	UÆ 0	KT •	* Z == (72	1 474 1 474	
ب	© • • • • · · · ·		XOO	208.	•C.g	>	E 🔿	~ □□	*O1	–ຕ⊢່	- X80 - X80 - X80	오모모
<	ドドら *1024	LLXL	יוטודי ו	uu uu `	\!L=		いルン	<u.u.=< td=""><td>· LL (</td><td>Y.LL L</td><td>⊔₩⊃ ►</td><td>œ.</td></u.u.=<>	· LL (Y.LL L	⊔ ₩⊃ ►	œ.
Ž	~~~									- 47		ய–ய
∑ α′	00	2,	33	04	95	90	7	8	60	01	=	N N N N N N N N N N N N N N N N N N N
T.	00	009	909	õ	6009	0	00	00	8	0,	0	-04
-	ru ra	9 4	• •	9	9	9	9	9	9	9	•	#*

```
LABEL SIMJLATION DE A PRESSURIZED WATER REACTOR
DUTPUT TIME, TAVE 500, 550, 174 ($ 20, 050, 1) TC ($ 500, 550, 1)
PAGE KYPLOT
DAGE KYPLOT
DAGE KYPLOT
DUJPUT TIME, POWER (0.2), 100, 1, 1, 2 AD (0.0)
DAGE HEIGHT = 5, WIDTH = 7
PAGE KYPLOT

WOUTPUT TIME, RHO! (-.0); WIDTH = 7
PAGE KYPLOT

WOUTPUT TIME, RHO! (-.0); WIDTH = 7

TIMER FINTIW= 500, JUTOEL = 0.1, PROEL = 5.0

END IN PUT

RESE TO THE FORMER THOUSE THE SEATOR

FILE SIMJLATION OF A PRESSURIZED WATER REACTOR

END IN PUT

RANGE RHO; RHOF; RHOT; RANGE (-.0); PROMER, LOAD

PAGE KYPLOT

FANGE RHO; RHOT; RHOT; RANGE (-.0); DAD (0.0)

LABEL SIMJLATION OF A PRESSURIZED WATER REACTOR

OUTPUT TIME, POMER (0.2); LOO, 1, LOO, 100, 0)

LABEL SIMJLATION OF A PRESSURIZED WATER REACTOR

OUTPUT TIME, POMER (0.2); LOO, 1, LOO, 100, 0)

LABEL SIMJLATION OF A PRESSURIZED WATER REACTOR

OUTPUT TIME, POMER (0.2); LOO, 1, LOO, 100, 0)

LABEL SIMJLATION OF A PRESSURIZED WATER REACTOR

OUTPUT TIME, POMER (0.2); LOO, 1, LOO, 100, 0)

LABEL SIMJLATION OF A PRESSURIZE WAS ENTREDED TO THE SERVICE OF THE SERVICE OF
```

**
OUTPUT TIME, RHO(--01, 01), RHDE(-.01, 01), RHOT(-.01), RHOX(-.01)
LABEL
LABEL
LABEL
PAGE HEIGHT = 5, WI)TH = 7 TRANSIENT
PAGE XYPLOT = 5, WI)TH = 7
TIMER FINTIM=5.4E+3, JUTOEL=1.0, PROEL=60.0
END
INPUT
END INPUT
STOP
END JOB

Charles Sales

APPENDIX G

INTERFACE PROGRAM LISTING

ETYPEDUT ERDR ETYPEDUT ON VSET ROYMSG DFF GLOBAL T TEK IB SYS. IB SSPLIB GLOBAL T TEK IB SYS. IB SSPLIB CP SET LINELN 80 FILEDEF OI DSK INITA. DATA LOAD INTACT ECSMP 21 CORE CSMP PI ENDI CSMP PI COMBINE PLANTS CSMP 21 PLANTS CSMP PI ENDI CSMP PI COMBINE PLANTS CSMP 21 PLANTS CSMP PI ENDI CSMP PI COMBINE PLANTS CSMP 21 PLANTS CSMP PI ENDI SMP PI SUBMIT PLANT CSMP 21 PLANTS SMP PI ENDI SMP PI SUBMIT PLANT CSMP 21 PLANTS SMP PI ENDI

LIST OF REFERENCES

- Schultz, M. A., <u>Control of Nuclear Reactors and Power Plants</u>, 2nd ed., McGraw-Hill, 1961.
- 2. Bell, G. I. and Glasstone S., <u>Nuclear Reactor Theory</u>, 1st ed., Van Nostrand, 1963.
- 3. Radkowsky, A. (ed.), <u>Naval Reactors Physics Handbook</u>, vol. I, USAEC, 1964.
- 4. Henry, A. F., Nuclear Reactor Analysis, 1st ed., MIT, 1975.
- 5. Duderstadt, J. J. and Hamilton, L. J., <u>Nuclear Reactor Analysis</u>, lst ed., Wiley, 1976.
- 6. The Shippingport Pressurized Water Reactor, personnel of the Naval Reactors Branch, AEC, and Westinghouse Electric, Addison-Wesley, 1958.

INITIAL DISTRIBUTION LIST

		No.	Copies
1.	Defense Technical Information Center Cameron Station Alexandria, Virginia 22314		2
2.	Library, Code 0142 Naval Postgraduate School Monterey, California 93940		2
3.	Department Chairman, Code 69 Department of Mechanical Engineering Naval Postgraduate School Monterey, California 93940		1
4.	Professor Paul J. Marto, Code 69Mx Department of Mechanical Engineering Naval Postgraduate School Monterey, California 93940		1
5.	Professor Gilles Cantin, Code 69Ci Department of Mechanical Engineering Naval Postgraduate School Monterey, California 93940		1
6.	CDR C. J. Garritson, USN, Code 34 Naval Engineering Curricular Officer Naval Postgraduate School Monterey, California 93940		1
7.	LCDR G. G. Heath, USN Department Head Class 67 SWOSCOLCOM, Bldg. 446 Newport, Rhode Island 02840		1

